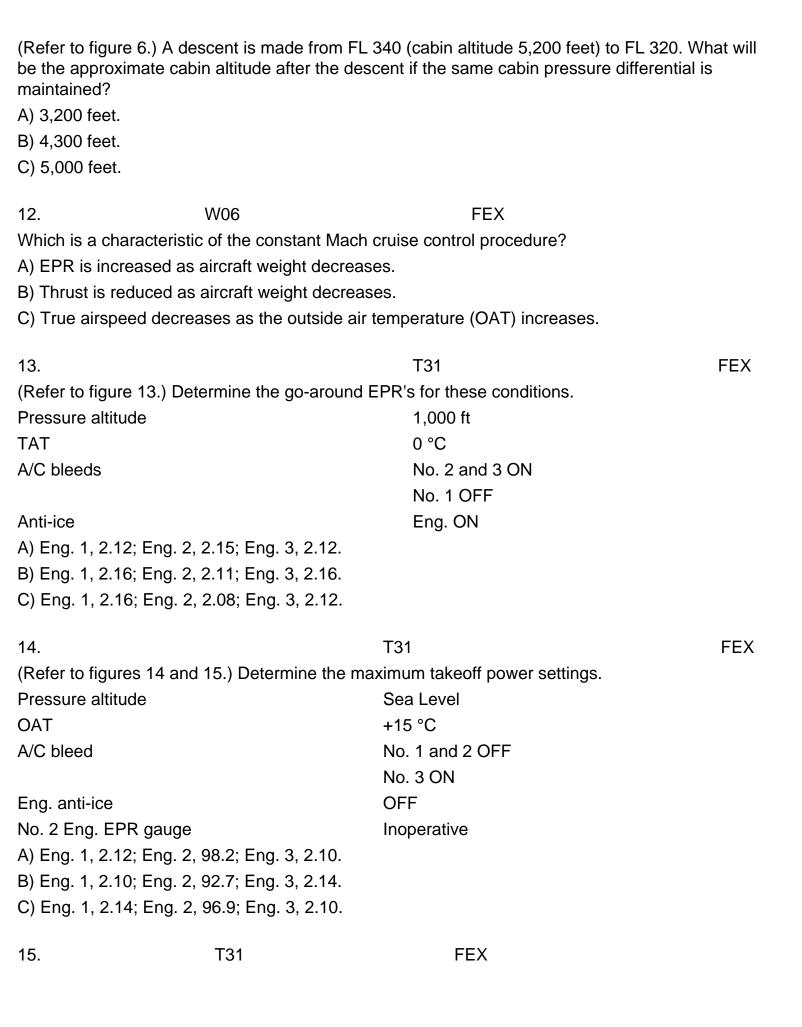
| 06/09/2006 Bank: (Flight Engineer) Airman Knowledge Test Question Bank  |                            |  |  |
|---|----------------------------|--|--|
| The FAA computer-assisted testing system is supported by a series of supplement publications. These publications, available through several aviation publishers, include the graphics, legends, and maps that are needed to successfully respond to certain test items. Use the following URL to download a complete list of associated supplement books: <a href="http://av-info.faa.gov/data/computertesting/supplements.pdf">http://av-info.faa.gov/data/computertesting/supplements.pdf</a> |                            |  |  |
| 1.  | T33                        | FEX  |  |
| Transonic   | airspeeds are considered   | l to be from   |  |
| A) Mach 0   |                            |  |  |
| B) Mach 0   |                            |  |  |
| C) Mach 0   | .75 to 2.0.                |  |  |
| 2.  | T33                        | FEX  |  |
| Mach num  | ber is commonly defined    | as the   |  |
| A) ratio of   | true airspeed to the spee  | d of sound.  |  |
| B) ratio of   | equivalent airspeed to the | e speed of sound.  |  |
| C) speed of   | of sound under conditions  | of standard pressure and temperature.                          |  |
| 3.  | T33                        | FEX  |  |
| The speed   | at which the airflow over  | the wing first reaches the speed of sound is known as the      |  |
| A) Reynold  | ds number.                 |  |  |
| B) transon  | ic index.                  |  |  |
| C) critical I   | Mach number.               |  |  |
| 4.  | T57                        | FEX  |  |
| The purpo   | se of sweeping wings bad   | ck approximately 30° to 35° is to                              |  |
| A) minimiz  | e dutch roll.              |  |  |
| B) reduce   | high-speed drag.           |  |  |
| C) provide  | aileron control when the   | root of the wing approaches the critical angle of attack.      |  |
| 5.  | T33                        | FEX  |  |
| Shock-indomay result  | -                          | occurring symmetrically near the wing root of a sweptback wing |  |
| A) severe   | porpoising due to an atte  | mpt to recover control while under reverse command.            |  |

| B) a high-spee  | ed stall and sudden pitch                        | up due to the center of pressure moving forward on the wing.   |
|-----------------|--|--|
| ,               | ving moment, due to the the the horizontal tail. | center of pressure moving aft on the wing and a decrease of  |
| 6.              | W06  | FEX  |
| An airplane is  | climbing at Mach .78. T                          | ne true airspeed will  |
| A) increase wi  | th altitude.                                     |  |
| B) increase as  | pressure decreases.                              |  |
| C) decrease a   | s the temperature decre                          | ases.  |
| 7.              | W07  | FEX  |
| Which is an ac  | dvantage of flying a jet a                       | high altitudes?  |
| A) Lower temp   | peratures increase engir                         | e efficiency.  |
| B) Thrust incre | eases as the density of t                        | ne air decreases.  |
| C) Lower engi   | ne RPM's will result in d                        | ecreased specific fuel consumption.  |
| 8.              | T30  | FEX  |
| When will the   | static air temperature ed                        | ual ram air temperature?   |
| A) When the a   | irplane is stationary.                           |  |
| B) At airplane  | speeds above Mach 0.3                            | 0.   |
| C) When the ra  | am air temperature is se                         | nsed by a Lewis Flush Bulb.  |
| 9.              | W07  | FEX  |
| Which factor is | s most significant when                          | letermining the optimum cruise altitude available?   |
| A) Winds aloft  | and temperature foreca                           | st.  |
| B) Fuel require | ement to climb to altitude                       | <b>).</b>  |
| C) Gross weig   | ht of the airplane at the                        | peginning of the cruise.   |
| 10.             | <b>S</b> 69                                      | FEX  |
| window and m    | ust decrease the cabin                           | cabin altitude of 8,000 feet experiences a cracked inner pressure differential from 8.6 to 5.0 PSI. To which altitude the same cabin altitude? |
| C) FL 230.      |  |  |
| 11.             | S69  | FEX  |



| es at a speed of Mach .82. Total fuel If Mach 1.0 is 595 knots, what has been the |
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|   |
|   |
|   |
| FEX   |
| for takeoff is ISA +34 °C. Which is the 0-foot pressure altitude airport?         |
|   |
|   |
| FEX   |
| for takeoff is ISA +34 °C. Which is the 00-foot pressure altitude airport?        |
| FEX   |
| is likely to form?  |
| FEX   |
| y to form in the engine inlet?  |
| FEX   |
|   |

| 21.                                | S78   | FEX  |
|------------------------------------|---|--|
| Why will the EPF                   | R indication increase falsel                            | y if the P <sub>t</sub> probe at the engine nose dome opening ices                                 |
| up?                                |   |  |
| A) The P <sub>t</sub> probe        | vent will act as a P <sub>s</sub> probe                 | •  |
| B) Anti-ice air pre                | essurizes the nose dome a                               | and the vent hole causing an increase in EPR.  |
| C) The reduced in pressure.        | inlet area causes pressure                              | to increase, and magnifies the influence of ram air  |
| 22.                                | S78   | FEX  |
| Why is engine an A) The air is too | •   | ed below an ambient temperature of +5 °F?  |
| B) Engine anti-ic                  | e will increase the intake t                            | emperature causing compressor ice to form.   |
|                                    |   | e engine or even cause the engine to fail completely.  |
| 23.                                | S24   | FEX  |
|                                    | t is equipped with heated in usually activated during f | nlet ducts and airfoil leading edges. When is this type of light?                                  |
| A) It is operated                  | continuously while in flight                            |  |
| B) At all times wi                 | hen the OAT is below free:                              | zing.  |
| C) Whenever icir                   | ng conditions are first enco                            | ountered or are expected to occur.   |
| 24.                                | S78   | FEX  |
| What will cause ticed closed?      | the EPR indication to decr                              | ease falsely if the P <sub>t</sub> probe at the engine nose dome is                                |
| A) The P <sub>t</sub> probe        | vent will act as a P <sub>s</sub> probe                 |  |
| •                                  | •   | e and the vent hole causing a decrease in EPR. to increase, and magnifies the influence of ram air |
| 25.                                | T34   | FEX  |
| Which statement                    | t is correct when applying                              | iquid rain repellent?  |
| A) Begin applica                   | tion as soon as rain begins                             | s, to form a barrier between the rain and the windshield.  |
| B) Apply rain rep                  | ellent first, then activate th                          | ne windshield wipers to spread the repellent.  |
| C) The number of                   | of times the repellent is app                           | olied is determined by the intensity of the rain.  |
| 26.                                | S66   | FEX  |
|                                    |   | generator is being driven, but there is no field excitation?                                       |
| A) Real voltage.                   | ,   | 5  |

| B) Residual voltage.   |  |                           |
|--|--|---------------------------|
| C) Reactive voltage.   |  |                           |
| <ul><li>27.</li><li>The purpose of a constant</li><li>A) control field strength.</li><li>B) regulate generator volt</li><li>C) maintain a uniform free</li></ul> |  | FEX<br>or is to           |
| <ul><li>28.</li><li>The purpose of a KVAR r</li><li>A) display the generator f</li><li>B) measure the work bein</li><li>C) indicate how hard the</li></ul>       | requency.  | FEX the power being used. |
| A) Open phase, underext<br>B) Undervoltage, differen   | T72 etions of an ac generator control citation, and overvoltage. tial fault, and manual paralleling and bus-tie circuit-breaker auto |                           |
| <ul><li>30.</li><li>The purpose of a KW me</li><li>A) display the generator f</li><li>B) measure the work beir</li><li>C) indicate how hard the</li></ul>        | requency.  | FEX the power being used. |
| <ul><li>31.</li><li>How are airplane ac gene</li><li>A) Volts.</li><li>B) Kilowatts (KW).</li><li>C) Kilovolt-amps (KVA).</li></ul>                              | S66<br>erators rated?  | FEX                       |
|  |  |                           |
| o, passing an unough co  | omig constitut contain a volunte   | , or ronnycraint.         |

| drop of air |
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| 39.                               | S27                          | FEX   |
|-----------------------------------|------------------------------|---|
| (Refer to figure does this indica |                              | om port 3, and a yellow disc is missing from port 1. What |
| A) Both bottles                   | have been normally dischar   | rged, but into different engines.                         |
| B) Bottle 1 has                   | been normally discharged a   | and bottle 2 has been thermally discharged.               |
| C) Bottle 2 has                   | been normally discharged a   | and bottle 1 has been thermally discharged.               |
| 40.                               | T35                          | FEX   |
| Airplanes equip during            | ped with both inboard and o  | outboard ailerons normally use the outboard ailerons only |
| A) low-speed or                   | perations.                   |   |
| B) high-speed of                  | perations.                   |   |
| C) low-altitude                   | operations.                  |   |
| 41.                               | T57                          | FEX   |
| Why do some a speeds only?        | irplanes equipped with inbo  | pard/outboard ailerons use the outboards for slow-flight  |
| A) Aerodynamic                    | c loads on the outboard aile | rons tend to twist the wingtips at high speeds.           |
| B) Increased su                   | ırface area provides greater | r controllability with flap extension.                    |
| C) Locking out                    | the outboard ailerons in hig | h speed flight provides variable flight control feel.     |
| 42.                               | S55                          | FEX   |
| How does an ai                    | ileron balance panel functio | n?  |
| A) A weight is in                 | nstalled ahead of the hinge  | line to counteract flight loads.                          |
| B) The aileron i                  | s extended ahead of the hir  | nge line so the airstream will help move the surface.     |
| C) Pressure cha                   | anges created by the aileron | n deflect a hinged panel in a compartment ahead of the    |
| 43.                               | T45                          | FEX   |
| When are outbo                    | oard ailerons normally used  | ?   |
| A) Low-speed f                    | light.                       |   |
| B) High-speed f                   | flight.                      |   |
| C) Low-speed a                    | and high-speed flight.       |   |
| 44.                               | T45                          | FEX   |
| When are inboa                    | ard ailerons normally used?  |   |
| A) Low-speed f                    | light only.                  |   |
| B) High-speed f                   | flight only.                 |   |
| C) Low-speed a                    | and high-speed flight.       |   |

| 45.                            | T34                         | FEX   |
|--------------------------------|-----------------------------|---|
| The use of a slo<br>because it | t in the leading edge of t  | he wing enables an airplane to land at a slower speed         |
| A) changes the                 | camber of the wing.         |   |
| B) delays the st               | all to a higher angle of at | tack.   |
| C) decelerates t               | the upper surface bound     | ary layer air.  |
| 46.                            | S55                         | FEX   |
| A purpose of wi                | ng mounted vortex gene      | rators is to  |
| A) prevent shoc                | k induced separation of     | air from the wing.  |
| B) increase the                | onset of drag divergence    | e and aid in aileron effectiveness at high speed.             |
| C) break the air               | flow over the wing so the   | stall will progress from the root out to the tip of the wing. |
| 47.                            | S22                         | FEX   |
| The purpose of                 | vortex generators mount     | ed on the vertical fin upstream of the rudder is to           |
| A) decrease dra                | ig at slow airspeeds.       |   |
| B) maintain rudo               | der effectiveness at high   | speed.  |
| C) prevent flow                | separation over the rudo    | er during extreme angles of yaw.                              |
| 48.                            | T45                         | FEX   |
| A purpose of lea               | ading edge slats on high    | performance wings is to                                       |
| A) decrease lift               | at relatively slow speeds   |   |
| B) improve ailer               | on control during low an    | gles of attack.   |
| C) direct air fror             | n the high-pressure area    | under the leading edge along the top of the wing.             |
| 49.                            | S55                         | FEX   |
| Which direction                | from the primary control    | surface does an antiservo tab move?                           |
| A) Same direction              | on.                         |   |
| B) Opposite dire               | ection.                     |   |
| C) Remains fixe                | ed for all positions.       |   |
| 50.                            | <b>S</b> 55                 | FEX   |
| Which direction                | from the primary control    | surface does a servo tab move?                                |
| A) Same direction              | on.                         |   |
| B) Opposite dire               | ection.                     |   |
| C) Remains fixe                | ed for all positions.       |   |

| 51.                | T34                            | FEX  |     |
|--------------------|--------------------------------|--|-----|
| What is a disadv   | antage of wing mounted vo      | rtex generators?                                       |     |
| A) Drag is increa  | ased slightly at slow airspee  | ds.  |     |
| B) Parasite drag   | increases significantly at hi  | gh airspeeds.  |     |
| ,                  | ,                              | ex generators increases control surface buffet.        |     |
| ,                  | •                              | S  |     |
| 52.                | T45                            | FEX  |     |
| A purpose of lea   | ding edge slats on high per    | formance wings is to                                   |     |
| A) increase lift a | t relative slow speeds.        |  |     |
| B) improve ailer   | on control during low angles   | of attack.   |     |
| C) direct air from | n the low-pressure area und    | er the leading edge along the top of the wing.         |     |
| 53.                | T45                            | FEX  |     |
| The purpose of a   | a servo tab is to              |  |     |
| A) move the fligh  | nt controls in the event of ma | anual reversion.                                       |     |
| B) reduce contro   | ol forces by deflecting in the | proper direction to move a primary flight control.     |     |
| C) prevent a con   | ntrol surface from moving to   | a full deflection position due to aerodynamic forces.  |     |
| 54.                | T45                            | FEX  |     |
| A purpose of flig  |                                |  |     |
|                    | camber of the wing.            |  |     |
| •                  | hout increasing airspeed.      |  |     |
| •                  | over the top of the wing at h  | aigh angles of attack                                  |     |
| o) anot annow      | over the top of the wing at i  | ight angles of attack.                                 |     |
| 55.                | T45                            | FEX  |     |
| The purpose of a   | an antiservo tab is to         |  |     |
| A) move the fligh  | nt controls in the event of ma | anual reversion.                                       |     |
| B) reduce contro   | ol forces by deflecting in the | proper direction to move a primary flight control.     |     |
| C) prevent a cor   | ntrol surface from moving to   | a full deflection position due to aerodynamic forces.  |     |
| 56.                | S67                            | FEX  |     |
| Total air tempera  | ature is equal to              |  |     |
| A) OAT correcte    | •                              |  |     |
| •                  |                                | from adiabatic compression of the boundary layer.      |     |
|                    |                                | actor of the temperature sensor is equal to 100 percen | ıt. |
| 57.                | S68                            | FEX  |     |
|                    | ples of wide cut turbine fuel  |  |     |
|                    | 1                              | -  |     |

| A) Jet B and JP-   | 4.                           |   |
|--------------------|------------------------------|---|
| B) Jet A and JP-   | 4.                           |   |
| C) Jet A and Jet   | A-1.                         |   |
| 58.                | S04                          | FEX   |
| What is the diffe  | rence between turbine fue    | el Jet A and Jet A-1?                                   |
| A) Jet A is made   | for operation at extremel    | y low temperatures.                                     |
| •                  | de for operation at extrem   |   |
| •                  | •                            | while Jet A-1 is formulated for the newest aircraft.    |
| 59.                | S68                          | FEX   |
| Which indicator i  | is used to determine if cor  | nditions are conducive to formation of ice in the fuel? |
| A) OAT.            |                              |   |
| B) Fuel temperat   | ture.                        |   |
| -                  | e warning system.            |   |
| 60.                | S68                          | FEX   |
| Which is a mean    | s of controlling the fuel te | mperature on turbojet-powered airplanes?                |
| A) Electrically he | eated fuel filters.          |   |
| B) Engine bleed    | air routed to a heat excha   | anger.  |
| C) Fuel filters he | eated by engine lubricating  | ı oil.  |
| 61.                | S68                          | FEX   |
| The purpose of a   | a fuel temperature indicato  | or is to  |
| A) determine if fu | uel temperatures are cond    | ducive to ice crystal formation.                        |
| B) determine if fu | uel temperatures are beyo    | and limits for proper combustion.                       |
| C) correct fuel qu | uantity indicator readings   | when the temperature is not standard.                   |
| 62.                | T24                          | FEX   |
| Which is the mos   | st critical parameter for a  | turbine engine during starting?                         |
| A) Oil pressure.   |                              |   |
| B) EGT.            |                              |   |
| C) Starter engag   | gement time.                 |   |
| 63.                | S76                          | FEX   |
| Fuel heaters sho   | ould not be operated on ta   | keoff, approach, or go-around because the               |
| A) EPR will decr   | ease significantly.          |   |
| B) engine may fl   | ameout from fuel vaporiza    | ation.  |

| C) oil temperature will inc   | crease significantly as fuel temper  | eratures rise within the oil cooler. |
|---|--|--------------------------------------|
| 64. An advantage of Skydrol A) is resistant to water co B) has a wide operating t C) is compatible with veg   | ontamination.  | FEX                                  |
| •   |  |                                      |
| B) reduce the tendency to   | S64 n tires is to now or ice covered runways. o hydroplane on wet runways. away from the engine intakes. | FEX                                  |
| <ul><li>67.</li><li>Where should the chines</li><li>A) One on each side of the</li><li>B) On the inside of the tir</li><li>C) On the outside of the tir</li></ul> | es only.   | FEX<br>I installation?               |
| 68. The main purpose of a brack A) provide pneumatic brack B) reduce the hydraulic pC) decrease the volume of   | ıking.   | FEX                                  |
| 69. One purpose of a brake of A) provide emergency brack B) increase the hydraulic C) increase the volume of  | aking.   | FEX                                  |

| 70.                                     | T02  | FEX  |
|---|--|--|
| The speed (RPM referred to as           | or percent) of the low                           | pressure compressor of a dual compressor engine is   |
| A) N <sub>1</sub> .                     |  |  |
| B) N <sub>2</sub> .                     |  |  |
| C) N <sub>3</sub> .                     |  |  |
| 71.                                     | S46  | FEX  |
|   | does engine pressure                             |  |
| A) Uncorrected co                       | ompressor inlet pressu                           | re and turbine discharge pressure.   |
|   | -  | ected for inlet duct loss and turbine discharge total pressure. rected for temperature and turbine discharge total pressure.   |
| 72.                                     | S46  | FEX  |
| A) The range beto<br>B) Decreased aer   | ween high and low spe<br>rodynamic drag is not a | um altitude for most airplanes during long range flights? ed Mach buffet decreases rapidly above this altitude. as great as the decrease in engine thrust above this altitude. ag from high angles of attack is greater than the increase in |
| 73.                                     | <b>S</b> 79                                      | FEX  |
| A) Drive shaft she<br>B) Sprag clutch a | ear point.<br>ssembly.                           | engine from driving a pneumatic starter to burst speed?  anes which causes choking.  |
| 74.                                     | <b>S</b> 79                                      | FEX  |
| A) engine RPM st<br>B) air manifold pr  | •  | •  |
| 75.                                     | S46  | FEX  |
|   | w compressor system                              | the first stage turbine drives the   |
| B) N <sub>2</sub> compresso             | r.   |  |
|   |  |  |

| C) N <sub>1</sub> and N <sub>2</sub> compresso   | ors.  |   |
|--|---|---|
| A) hot section burnout of B) compressor tempera  | ture limits from being exceeded.                              | ·   |
| A) The turbine case cool B) Rapid cooling of the   | ols faster and may shrink down a compressor section may cause |   |
| <ul> <li>78.</li> <li>The speed (RPM or per referred to as</li> <li>A) N<sub>1</sub>.</li> <li>B) N<sub>2</sub>.</li> <li>C) N<sub>3</sub>.</li> </ul> | S46 recent) of the high pressure comp                         | FEX ressor of a dual compressor engine is |
| <ul><li>79.</li><li>(Refer to figure 7.) The</li><li>A) Location 3.</li><li>B) Location 4.</li><li>C) Location 6.</li></ul>                            | T03<br>internal engine temperature will                       | FEX be the lowest at which location?      |
| <ul><li>80.</li><li>(Refer to figure 7.) The</li><li>A) Location 4.</li><li>B) Location 5.</li><li>C) Location 6.</li></ul>                            | T03<br>internal engine pressure will be                       | FEX the lowest at which location?         |
| 81. (Refer to figure 7.) The   | T03 internal engine pressure will be                          | FEX the highest at which location?        |

| A) Location 4.  |                              |   |
|---|------------------------------|---|
| B) Location 5.  |                              |   |
| C) Location 6.  |                              |   |
|   |                              |   |
| 82.   | T19                          | FEX   |
| The primary purpo                                       | ose of an oil-to-fuel heat e | exchanger is to                                       |
| A) cool the oil.  |                              |   |
| B) heat the fuel.                                       |                              |   |
| C) decrease oil vis                                     | scosity.                     |   |
| 83.   | S46                          | FEX   |
| (Refer to figure 8.)                                    | ) Where is the engine hig    | h pressure compressor located?                        |
| A) Location 1.  | ,                            |   |
| B) Location 2.  |                              |   |
| C) Location 3.  |                              |   |
|   |                              |   |
| 84.   | S46                          | FEX   |
|   |                              | w pressure compressor speed of a triple-spool engine? |
| A) The compresso  | or will speed up as altitud  | e is increased.                                       |
| B) The fuel contro                                      | ol will maintain a constant  | compressor speed.                                     |
| C) The compresso  | or speed will decrease as    | the atmosphere becomes less dense.                    |
| 85.   | S73                          | FEX   |
|   | ) Power to drive the N₁ co   | ompressor comes from the turbine at which location?   |
|   | , 1                          | •   |
| A) Location 5.  |                              |   |
| <ul><li>B) Location 8.</li><li>C) Location 9.</li></ul> |                              |   |
| C) Location 9.  |                              |   |
| 86.   | S81                          | FEX   |
| (Refer to figure 10                                     | ).) What is the indicated s  | peed of the engine low pressure compressor?           |
| A) 94.1 percent.  |                              |   |
| B) 96.5 percent.  |                              |   |
| C) 97.0 percent.  |                              |   |
| 87.   | S46                          | FEX   |
|   |                              | y-pressure turbine located?                           |
| A) Location 2.  | ,                            | •   |
| B) Location 3.  |                              |   |

| C) Location 4.                    |   |   |
|-----------------------------------|---|---|
| 88.                               | S46   | FEX   |
| On most engines                   | s ram recovery occurs abo                             | ve  |
| A) 60 knots.                      |   |   |
| B) 100 knots.                     |   |   |
| C) 140 knots.                     |   |   |
| 89.                               | S72   | FEX   |
| The function of t                 | he nozzle diaphragm locat                             | ted on the upstream side of the turbine wheel is to           |
| A) increase the p                 | oressure of the exhaust ma                            | ass.  |
| B) increase the                   | velocity of the heated gase                           | es flowing past the nozzle diaphragm.                         |
| C) decrease the                   | velocity of the heated gas                            | es flowing past the nozzle diaphragm.                         |
| 90.                               | S72   | FEX   |
| The purpose of t                  | the diffuser section in a tur                         | bine engine is to   |
| A) convert press                  | ure to velocity.                                      |   |
| B) increase pres                  | sure and reduce velocity.                             |   |
| C) reduce press                   | ure and increase velocity.                            |   |
| 91.                               | S83   | FEX   |
|                                   | tting an engine down by tu<br>uring emergencies only? | rning the boost pumps off and closing the airplane's fuel     |
| A) During a subs                  | sequent start the engine di                           | iven fuel pump may cavitate causing a hung start.             |
| B) The fuel syste lubricated.     | em's service life will be ded                         | creased because fuel wetted components will not be            |
| C) The 30 to 60 too long for norm |   | ar the fuel lines from the firewall to the fuel controller is |
| 92.                               | <b>S</b> 49   | FEX   |
| Which compone operations?         | nts in a turbine engine aid                           | in stabilization of the compressor during low power           |
| A) Stator vanes.                  |   |   |
| B) Bleed air valv                 | es.   |   |
| C) Pressurization                 | n and dump valves.                                    |   |
| 93.                               | S83   | FEX   |
| If a turbine engir                | ne catches fire internally du                         | uring the start cycle, what initial action should be taken?   |
| A) Secure all sw                  | itches and leave the airpla                           | ne.   |

| •  | el and continue motorin<br>2 into the inlet duct whil    | g the starter. e continuing to motor the engine.                           |
|--|--|--|
| 94.<br>(Refer to figure 7.<br>A) Location 3.<br>B) Location 4.<br>C) Location 6. | T03<br>) The internal engine te                          | FEX emperature will be the highest at which location?                      |
| 95.<br>Which engine inst<br>A) N <sub>2</sub> .<br>B) EPR.<br>C) EGT or TIT.     | S81<br>trument is most likely to                         | FEX show an elevated reading if the turbine wheel is damaged               |
| A) Bleed valve stuB) Ice in the inlet  | uck open.  | FEX PR indication, in flight or on the ground? es.                         |
| 97. What condition wisettings? A) FOD. B) Bleed valve stu                        | uck open.  | FEX<br>/ <sub>f</sub> , and RPM with a normal EPR indication, at all power |
| A) Traffic priority i<br>B) Emergency ha   |  |  |
|  | T50<br>does a Mach meter pre<br>plane true airspeed to t |  |

| B) The ratio of airplane indicated airspee          | d to the speed of sound.                                      |   |
|---|---|---|
| C) The ratio of airplane equivalent airspe          | ed, corrected for installation error, to the speed of sound.  |   |
| 100. D20  | FEX   |   |
| Each crewmember shall have available for            | or individual use on each flight a                            |   |
| A) quick-donning type oxygen mask.                  |   |   |
| B) flashlight in good working order.                |   |   |
| C) hand fire extinguisher suitable for extinguisher | nguishing Class A, B, and C fires.                            |   |
| 101. A07  | FEX   |   |
| main fuel control? Forty-five minutes at            | main for turbine-powered airplanes after jettisoning with the | ţ |
| A) 75 percent maximum continuous power              |   |   |
| B) maximum continuous power with the c              | •   |   |
| C) maximum range speed after climbing               | from sea level to 10,000 feet.                                |   |
| 102. J11  | FEX   |   |
| What altitude information is transmitted b          | y MODE C of the transponder?                                  |   |
| A) Pilot's indicated altitude.                      |   |   |
| B) Altitude in 10-foot increments.                  |   |   |
| C) Altitude without barometric pressure c           | correction.   |   |
| 103. J13  | FEX   |   |
| What is the significance of a flashing red          | ATCT light gun signal?  |   |
| A) Vehicles or personnel should clear the           | e taxiway.  |   |
| B) Aircraft in flight should exercise extren        | ne caution.   |   |
| C) Aircraft on the ground should return to          | their starting point on the airport.                          |   |
| 104. I29  | FEX   |   |
| An in-flight condition necessary for struct         | ural icing to form is   |   |
| A) visible moisture.                                | -   |   |
| B) stratiform clouds.                               |   |   |
| C) cirrostratus clouds.                             |   |   |
| 105.  | H124 FEX  |   |
| Before a cargo change is made, the follow           |   |   |
| Aircraft weight                                     | 250,000 lb  |   |
| CG  | 30.5 percent of MAC   |   |
|   |   |   |

| Length of MAC                                 | ation 763.0 to 1035.3     |  |  |
|---|---------------------------|--|--|
| -   | is removed from an a      | verage location of Station 1010.0, what is the new     |  |
| CG relative to MAC?                           |                           |  |  |
| A) 27.5 percent.                              |                           |  |  |
| B) 29.3 percent.                              |                           |  |  |
| C) 31.7 percent.                              |                           |  |  |
| 106.  | H124                      | FEX  |  |
| The gross weight of the a                     | airplane is 155,000 po    | ounds. How much weight must be moved from Station      |  |
| 1028.0 to Station 582.0 to                    | o move the CG forwa       | rd 1.2 inches?   |  |
| A) 352 pounds.                                |                           |  |  |
| B) 418 pounds.                                |                           |  |  |
| C) 516 pounds.                                |                           |  |  |
| 107.  | H124                      | FEX  |  |
| What minimum weight of CG within limits?      | cargo must be shifted     | d from the aft to the forward compartment to bring the |  |
| Total weight                                  | 165,000 lb                |  |  |
| MAC   | Station 860.2             | to 1040.9  |  |
| CG  | 34.0 percent of MAC       |  |  |
| Aft CG limit                                  | 32.0 percent              | of MAC   |  |
| Cargo centroids:                              |                           |  |  |
|   | Fwd                       | 582 in   |  |
|   | Aft                       | 1028 in  |  |
| A) 740 pounds.                                |                           |  |  |
| B) 1,032 pounds.                              |                           |  |  |
| C) 1,338 pounds.                              |                           |  |  |
| 108.  | W12                       | FEX  |  |
| Compared to a no-wind of takeoff performance? | condition, what effect    | would a 20 knot headwind component have on             |  |
| A) The effect of wind on i                    | initial acceleration will | result in a longer takeoff roll.                       |  |
| B) The airplane will reach                    | n critical engine failure | e indicated airspeed at a lower groundspeed.           |  |
| C) Critical engine failure condition.         | speed and actual gro      | undspeed will be the same as in a zero-wind            |  |
| 109.  | W04                       | FEX  |  |
| The true airspeed at which                    | ch an airplane stalls v   | aries with   |  |

A) load factor and angle of attack.

| B) load factor, weight, a C) density altitude, wei                                   | and density altitude.<br>ght, and angle of attack.  |  |        |
|--|---|--|--------|
| 110.   | T35   | FEX  |        |
| An airplane is in equilib  | orium when  |  |        |
| A) there are no acceler  | ations and the airplane   | continues in steady flight.                      |        |
| B) the airplane is distur  | rbed from its flightpath a  | nd it will return without control use.           |        |
| C) the airplane has nei  | ther the tendency to con  | tinue or return from disturbance displaceme      | nt.    |
| 111.   | W05   | FEX  |        |
| A) increases as weight<br>B) remains constant re                                     | ich produces the highest<br>or altitude is increased.<br>gardless of weight or alti<br>s altitude is changed, but |  |        |
| 112.   | T33   | FEX  |        |
| What effect will decrea  | sing air density have on  | lift and drag?                                   |        |
| A) Lift and drag will ded  | •   | -  |        |
| B) Lift will increase and  | d drag will decrease.   |  |        |
| C) Lift will decrease an   | d drag will increase.   |  |        |
| 113.   | W02   | FEX  |        |
| True airspeed is deterr  | nined by correcting   |  |        |
| A) indicated airspeed for  | or density altitude.  |  |        |
| B) calibrated airspeed   | for nonstandard tempera   | ature and altitude.                              |        |
| C) equivalent airspeed   | for the air-density variat  | tion from the standard value at sea level.       |        |
| 114.   | S69   | FEX  |        |
| (Refer to figure 6.) The<br>What is the differential<br>A) 4.44 PSI.<br>B) 4.71 PSI. | -   | is 6,000 feet and the airplane altitude is FL 1  | 180.   |
| C) 5.46 PSI.   |   |  |        |
| 115.   | S69   | FEX  |        |
| (Refer to figure 6.) The<br>What is the cabin altitu                                 |   | tial is 5.46 PSI and the airplane altitude is FL | _ 200. |
| A) 3,200 feet.   |   |  |        |

| B) 4,400 feet.<br>C) 5,000 feet.  |  |                           |                     |
|---|--|---------------------------|---------------------|
| 116. T3<br>(Refer to figure 30.) How much fue<br>A) 710 pounds.<br>B) 2,917 pounds.<br>C) 3,294 pounds.   |  | FEX<br>under operating co | nditions No. 1?     |
| 117. T3   | 31   | FEX                       |                     |
| (Refer to figures 22 and 23.) What<br>A) 234 BMEP.<br>B) 204 BMEP.<br>C) 59.5 inches manifold pressure.   | is the takeoff power sett                                | ting under operating      | g conditions No. 1? |
| 118.  | T31  | FEX                       |                     |
| (Refer to figure 29.) How much fue  | el remains after operating                               | g under these condi       | tions?              |
| Number of engines   | 4  |                           |                     |
| Beginning total weight  | 95,720 lb  |                           |                     |
| Zero fuel weight  | 64,850 lb  |                           |                     |
| ВНР   | Pressure Alt.  | Spark                     | Time                |
| 1,200   | 17,000 ft  | T/O & CL                  | 18 min              |
| 1,100   | 19,000 ft  | Cruise                    | 1 hr 20 min         |
| 1,000   | 19,000 ft  | Cruise                    | 1 hr 10 min         |
| A) 2,040 gallons.   |  |                           |                     |
| B) 3,874 gallons.   |  |                           |                     |
| C) 5,145 gallons.   |  |                           |                     |
| 119. W0 <sup>-</sup>  | 7  | FEX                       |                     |
| For an airplane with a given gross between fuel flow, temperature, an A) both temperature and altitude a B) both temperature and altitude a C) temperature is increased and a | nd altitude? Fuel flow is he re decreased. re increased. | •                         | he relationship     |
| 120. T1<br>Which is an indication of carbureto<br>A) Decrease in propeller RPM.   |  | FEX                       |                     |

| B) Manifold pressure (MAP) dro                                   | pp.                             |  |
|--|---------------------------------|--|
| C) Backfiring, which is caused by                                | oy a rich mixture.              |  |
| 121.   | T12                             | FEX                                    |
| Which atmospheric condition is                                   |                                 |  |
| A) High aridity and freezing tem                                 |                                 | o loo with partial power.              |
| B) Rain with the ambient tempe                                   |                                 | freezina                               |
| C) High humidity and an ambier                                   | _                               | •                                      |
| C) Flight humbles  | iii teiriperature between 50 -i | and oo i.                              |
| 122.   | T66                             | FEX                                    |
| What are the characteristics of                                  | the electrolyte in a nickel-cad | lmium battery?                         |
| A) Noncorrosive.   |                                 |  |
| B) Much like household lye and                                   | will cause severe burns.        |  |
| C) Harmless compared to the e                                    | lectrolyte in a lead-acid batte | ery.                                   |
|  |                                 |  |
| 123.   | T66                             | FEX                                    |
| What type of electrolyte is conta                                | ained in a lead-acid battery?   |  |
| A) Boric acid.   |                                 |  |
| B) Sulfuric acid.  |                                 |  |
| C) Potassium hydroxide.  |                                 |  |
| 124.   | S66                             | FEX                                    |
|  |                                 | enerator voltage is lower than battery |
| voltage?   |                                 |  |
| A) Voltage regulator.  |                                 |  |
| B) Paralleling circuit.  |                                 |  |
| C) Reverse-current relay.3/8/94                                  | chg ref from S51 to S66. P-     | 531.                                   |
| 125.   | S66                             | FEX                                    |
| The purpose of a reverse-curre                                   | nt relay is to                  |  |
| A) prevent one generator from (                                  | driving another generator.      |  |
| B) increase or decrease the vol                                  | tage of a generator so it carri | ies its share.                         |
| C) disconnect the generator from voltage.3/8/94 Chg ref. from S5 | _                               | ator voltage drops below battery       |
| 126.   | T75                             | FEX                                    |
| What is a feature of an electron                                 | nagnetic-type circuit breaker?  | •                                      |
| A) It resets automatically.                                      |                                 |  |

| B) It may be reset immediate             | ly.                              |  |
|--|----------------------------------|--|
| C) It is necessary to wait a sh          | nort time before attempting a    | reset.                                   |
| 127.                                     | T71                              | FEX                                      |
| Which type voltage regulator<br>current? | uses a variable resistance el    | ement for controlling dc generator field |
| A) Vibrator.                             |                                  |  |
| B) Solid-state.                          |                                  |  |
| C) Carbon-pile.                          |                                  |  |
| 128.                                     | T75                              | FEX                                      |
| What is a feature of a nontrip           | -free circuit breaker?           |  |
| A) It is impossible to manuall           | y hold it in the closed positior | າ.                                       |
| B) Emergency circuits may b              | e powered by holding it in a c   | closed position.                         |
| C) More amperage than its ra             | ated capacity and heavy over     | loads can be carried for a short time.   |
| 129.                                     | T72                              | FEX                                      |
| What is a purpose of a rotary            | inverter?                        |  |
| A) Change 115 volts ac to 28             | 3 volts dc.                      |  |
| B) Convert dc to 115 volts, 4            | 00-Hz power.                     |  |
| C) Transform 26/29 volts dc              | to 115/200 volts dc.             |  |
| 130.                                     | T75                              | FEX                                      |
| How many spare electrical fu             | ises are required for use in fli | ght?                                     |
| A) One complete spare set.               |                                  |  |
| B) Three fuses of each size t            | hat is installed.                |  |
| C) 50 percent for each rating            | required but not less than on    | ne for a particular rating.              |
| 131.                                     | S66                              | FEX                                      |
| What is residual voltage?                |                                  |  |
| A) Voltage produced that is r            | not in phase with the current.   |  |
| B) Voltage stored in the gene            | erator exciter output windings   |  |
| C) Voltage produced by pern              | nanent magnets which starts      | the ac generator output.                 |
| 132.                                     | T66                              | FEX                                      |
| What will neutralize the elect           | rolyte from a lead-acid battery  | y?                                       |
| A) Soap and water.                       |                                  |  |
| R) Ricarbonate of soda                   |                                  |  |

| C) Boric acid, a solution o   | f acetic acid, lemon juice, or vind | egar.   |
|---|-------------------------------------|---|
| 133.  | T69                                 | FEX   |
| disconnecting the battery?  A) To prevent discharging  B) To prevent a spark fror                         | ?<br>g the battery.                 | rces be turned off before connecting or nent. |
| 134. What is the difference bet A) Relays have movable o B) Solenoids have movab C) Relays are used as me | le cores.                           | FEX   |
| 135.  | S31                                 | FEX   |
| A) Open the outflow valve B) Close the outflow valve C) Increase the amount of                            | e slower.<br>e faster.              | surization controls be adjusted?              |
| 136.  | S31                                 | FEX   |
| A) Relieve a negative pres<br>B) Relieve all positive pres  |                                     |   |
| 137.  | S31                                 | FEX   |
| What component of a prest   | ssurization system prevents the     | cabin altitude from becoming higher than      |
| A) Cabin rate of descent of   |                                     |   |
| B) Negative pressure relie  |                                     |   |
| C) Compression ratio limit  | t switch.                           |   |
| 138.  | T30                                 | FEX   |
| Which best describes cab  | in differential pressure?           |   |
| A) The difference between   | n ambient and internal air pressu   | ure.  |
| B) The difference between   | n the cabin flight altitude pressu  | re and MSL pressure.                          |

| C) The difference between   | n the cabin pressu  | re controller setting and the actual cabin pressure.  |
|---|---------------------|---|
| 139.  | S31                 | FEX   |
| Which control systems for the controller to regulate A) Isobaric and differential | the outflow valve?  | ressurization use reference chamber air pressure within   |
| B) Unpressurized and pre  | essurized controls. |   |
| C) Ambient, differential, a   | and maximum differ  | ential.   |
| 140.  | <b>S</b> 69         | FEX   |
| What oxygen flow conditi position and the supply le                               |                     | ne oxygen regulator selector is placed in the emergency   |
| A) 100 percent oxygen av  | vailable on demand  | l <b>.</b>  |
| B) Continuous flow of dilu  | uted oxygen under   | positive pressure.  |
| C) Continuous flow of 100   | 0 percent oxygen u  | nder positive pressure.   |
| 141.  | T49                 | FEX   |
| What is one danger of an  | y oxygen leak?      |   |
|   | •                   | se combustible materials to burn intensely.  pidly and burn with greater intensity in oxygen rich |
| C) Any ignition source ma<br>surroundings in poorly ve                            |                     | losive oxygen which over a period of time saturates the   |
| 142.  | T49                 | FEX   |
| Why must oxygen not be  | permitted to come   | in contact with oil, grease, or solvents?   |
| A) Oxygen is highly flamr   | nable and will caus | e petroleum products to burn or explode.  |
| B) Oxygen contact with p  | etroleum products   | can cause spontaneous fires or explosions.  |
| C) Gaseous oxygen is ch explosive mixture.  | emically unstable a | and will combine with petroleum to form a highly  |
| 143.  | <b>S</b> 69         | FEX   |
| Which position should be cockpit?   | selected on the dil | uter-demand oxygen regulator if there is smoke in the   |
| A) Normal.  |                     |   |
| B) Emergency.   |                     |   |
| C) 100 percent.   |                     |   |
| 144.  | T34                 | FEX   |

| What direction should the ailero A) Left aileron up when the cont B) Right aileron down when the C) Right aileron down when the  | crol wheel is moved right.            |                               |
|--|---------------------------------------|-------------------------------|
| <ul><li>145.</li><li>Alcohol is added to the fluid in a</li><li>A) increase the octane.</li><li>B) provide increased cooling.</li><li>C) prevent freezing of the water</li></ul> |                                       | FEX                           |
| <ul><li>146.</li><li>Water-soluble oil is added to the A) prevent corrosion.</li><li>B) provide lubrication.</li><li>C) increase the viscosity of the analysis.</li></ul>        |                                       | FEX<br>for water injection to |
| <ul><li>147.</li><li>The purpose of water injection is</li><li>A) suppress detonation.</li><li>B) increase fuel economy.</li><li>C) prevent carburetor ice.</li></ul>            | T12<br>s to                           | FEX                           |
| <ul><li>148.</li><li>Which publication determines w</li><li>A) FAR Part 25.</li><li>B) The applicable manufacturer</li><li>C) AC 65-9A, Airframe and Pow</li></ul>               | 's manual.                            |                               |
| 149. T48 FEX A reason for using a crossfeed fuel system is to A) be able to purge any fuel tank. B) jettison fuel during emergencies. C) help maintain aircraft stability.       |                                       |                               |
| 150. One purpose of a fuel tank boos A) temperature.   | S04<br>st pump is to prevent vapor lo | FEX<br>ock caused by low      |

| B) altitude operation.   |                                 |                                      |  |
|--|---------------------------------|--------------------------------------|--|
| C) atmospheric pressure.   |                                 |                                      |  |
|  |                                 |                                      |  |
| 151.   | T10                             | FEX                                  |  |
| The purpose of the hopper tank   |                                 |                                      |  |
| A) permit a fast warmup of the   |                                 |                                      |  |
| B) collect sludge and particles f  | rom the oil in the event the oi | I filter becomes obstructed.         |  |
| C) provide engine oil to feather engine oil is pumped overboard  |                                 | oil line should break and all of the |  |
| 152.   | T46                             | FEX                                  |  |
| What action should be taken if a   | a hydraulic stationary connec   | tion has a static leak?              |  |
| A) Reduce the accumulator pre  |                                 |                                      |  |
| B) Notify maintenance to repair  | it.                             |                                      |  |
| C) Pressurize the system and p   |                                 |                                      |  |
| ,  | •                               |                                      |  |
| 153.   | S63                             | FEX                                  |  |
| What is a limitation of MIL-H-56   | 606?                            |                                      |  |
| A) Flammable.  |                                 |                                      |  |
| B) Chemically unstable.  |                                 |                                      |  |
| C) High viscosity change with te   | emperature.                     |                                      |  |
| 154.   | T46                             | FEX                                  |  |
| The purpose of a hydraulic accu  |                                 |                                      |  |
| A) store hydraulic fluid under pr  |                                 |                                      |  |
| B) collect hydraulic fluid from sy   |                                 |                                      |  |
| C) gather foam and extract the   |                                 | eservoir.                            |  |
| , 0  | J                               |                                      |  |
| 155.   | S63                             | FEX                                  |  |
| Pressure from an engine-driven   | n hydraulic pump may be regu    | ılated by a                          |  |
| A) system bypass valve.  |                                 |                                      |  |
| B) constant speed drive.   |                                 |                                      |  |
| C) in-line variable restrictor orifi   | ce.                             |                                      |  |
| 156.   | S63                             | FEX                                  |  |
|  |                                 |                                      |  |
| What is the difference between a hydraulic sequence valve and a priority valve?  A) Sequence valves are electrically actuated. |                                 |                                      |  |
| B) Mechanical contact opens a  | •                               |                                      |  |
|  | F. 13.1.9.                      |                                      |  |

| C) Priority valves are opened by hydraulic pressure.   |   |                                    |  |
|--|---|------------------------------------|--|
| 157.   | T46   | FEX                                |  |
| What will the hydraulic pressur monitor is on the air side of a d A) Zero.  B) System pressure.  C) Between one-third and one-   | liaphragm-type accumulator?                                       |                                    |  |
| 158.   | T46   | FEX                                |  |
| <ul><li>What type of gas may be used</li><li>A) Nitrogen.</li><li>B) Dry oxygen.</li><li>C) Carbon dioxide.</li></ul>  | to service hydraulic accumula                                     | ators?                             |  |
| <ul><li>159.</li><li>Which principle operates a hyd</li><li>A) Heat.</li><li>B) Electrical.</li><li>C) Differential pressure.</li></ul>  | T46<br>raulic fuse?   | FEX                                |  |
| 160.   | S63   | FEX                                |  |
| The purpose of pressurizing a A) provide an alternate source B) assure a positive feed of foa C) insure an adequate supply of  | of pressure in case of a hydra<br>m free fluid to the hydraulic p | oump at high altitudes.            |  |
| <ul><li>161.</li><li>Why should hydraulic fluid be f</li><li>A) Water in the fluid could free:</li><li>B) It assures a positive feed of</li><li>C) Contaminants may damage</li></ul> | ze.<br>foam free fluid to the hydraul                             | •                                  |  |
| 162. What is the significance of the A) Confirmation of fluid therma   | l expansion in the gear, flaps                                    | , or other systems.                |  |
| B) Evidence that contaminants  | may prevent components su   | ch as check valves from operating. |  |

C) Indication of overfilling and the fluid level of the hydraulic reservoir should be checked.

| 163.   | T46  | FEX  |
|--|--|--|
| One purpose of a hydra                       | aulic accumulator is to                              |  |
| A) compress hydraulic                        | fluid.   |  |
| B) absorb sudden pres                        |  |  |
| •  | from small system leaks.                             |  |
| , ,  |  |  |
| 164.   | S64  | FEX  |
| The purpose of fusible                       | plugs in the wheel is to pro                         | event  |
| A) tire blowout.                             |  |  |
| B) overservicing the tire                    | e.   |  |
| C) damage to the tire r                      | esulting from heat expansi                           | on.  |
| 165.   | T47  | FEX  |
| The purpose of fusible                       | plugs in aircraft wheels is                          | to   |
| A) prevent tire blowout                      | . •  |  |
| B) quickly deflate tires                     |  |  |
| C) protect the antiskid                      | •  |  |
| -, p   |  |  |
| 166.   | S26  | FEX  |
| One of the main gear t                       | ires has deflated as a resu                          | It of a thermal fuse melt. What does this mean?              |
| A) Heavy braking has danger of a tire blowou |  | ted a plastic fuse in the rim, and prevented the             |
| B) High tire temperatur the tire to deflate. | es have melted a fusible n                           | netal plug installed in the aircraft wheel and cause         |
| , .  | in the wheel well have cause and prevented damage to | sed the tire's temperature sensitive valve core to the wing. |
| 167.   | T46  | FEX  |
| What type of gas is not system?              | rmally used to service the                           | air-storage bottles of an emergency pneumatic                |
| A) Nitrogen.                                 |  |  |
| B) Dry oxygen.                               |  |  |
| C) Carbon dioxide.                           |  |  |
| 168.   | T46  | FEX  |
| Moisture in a pneumati                       |  |  |
| A) corrosion.                                | ,  |  |
| B) a variety of sounds                       | including banging, squeali                           | ng and chattering.   |

| C) return lines to freeze when                                   | the pressure of the air drops     | during actuation.                          |
|--|-----------------------------------|--|
| 169.   | S48                               | FEX  |
| Afterfiring is caused by   |                                   |  |
| A) the spontaneous combustic of the charge.                      | on of the unburned charge ah      | ead of the flame fronts after the ignition |
| B) charges of unburned fuel in causing an explosion in the ex    |                                   | air outside the exhaust and igniting,      |
| C) a mixture so lean that comb<br>the intake manifold when the i | -                                 | exhaust stroke ignites the contents of     |
| 170.   | S48                               | FEX  |
| What is detonation?  |                                   |  |
| A) The unburned fuel/air mixtuof the charge.                     | ire ahead of the flame fronts of  | explodes spontaneously after the ignition  |
| B) The charges of unburned fucausing an explosion in the ex      |                                   | h air outside the exhaust and ignite       |
| C) The mixture is so lean that the intake manifold ignite whe    | -                                 | at the exhaust stroke and the contents of  |
| 171.   | S48                               | FEX  |
| Backfiring is caused by  |                                   |  |
| A) fouled spark plugs, defective                                 | re fuel injection nozzles, or inc | correct valve clearances.                  |
| B) charges of unburned fuel in<br>causing an explosion in the ex | •                                 | air outside the exhaust and igniting,      |
| C) a mixture so lean that comb<br>the intake manifold when the i | -                                 | exhaust stroke ignites the contents of     |
| 172.   | S48                               | FEX  |
| Preignition is indicated by                                      |                                   |  |
| A) intermittent firing and low c                                 | ylinder temperatures.             |  |
| B) explosions from the exhaus                                    | st system with torching or afte   | rburning.                                  |
| C) engine roughness and a su                                     | idden increase in cylinder hea    | ad temperatures.                           |
| 173.   | T29                               | FEX  |
| f the line between the manifol<br>gauge will indicate            | d pressure gauge and the en       | gine induction system is broken, the       |
| A) ambient pressure.   |                                   |  |
| B) standard atmospheric press                                    | sure.                             |  |
| ?) high when operating at a m                                    | anifold pressure above atmos      | spheric pressure                           |

| 174.  | T29   | FEX   |        |
|---|---|---|--------|
| If the line between the indicate            | MAP gauge and the engine inc                                  | uction system has a leak, the gauge w                   | /ill   |
| A) ambient pressure.                        |   |   |        |
| B) 29.92 inches of MA                       | Р.  |   |        |
| C) low when operating                       | at a MAP above atmospheric p                                  | ressure.  |        |
| 175.  | S20   | FEX   |        |
| Which of the following system when operated | _   | the engine to backfire through the ind                  | uction |
| A) A clogged air inlet.                     |   |   |        |
| B) The idle speed set t                     | oo low.   |   |        |
| C) An excessively lean                      | mixture setting.  |   |        |
| 176.  | T10   | FEX   |        |
| Excessive oil in the lov                    | ver cylinder heads is an indicati                             | on of   |        |
| A) worn oil control ring                    | s.  |   |        |
| B) oil supply line bypas                    | ss valve failure.   |   |        |
| C) intercylinder drains                     | that are partially or completely                              | olocked.  |        |
| 177.  | S45   | FEX   |        |
| How are the cylinders                       | numbered in a double-row radia                                | ıl engine?  |        |
| A) Clockwise as viewe                       | d from the engine front.                                      |   |        |
| B) Clockwise as viewe                       | d from the accessory end.                                     |   |        |
| C) Counterclockwise a                       | s viewed from the accessory er                                | nd.   |        |
| 178.  | T11   | FEX   |        |
| The primary purpose of                      | f exhaust augmenters is to                                    |   |        |
| A) increase engine coo                      | oling.  |   |        |
| B) provide additional the                   | nrust.  |   |        |
| C) decrease exhaust b                       | ack pressure.   |   |        |
| 179.  | S20   | FEX   |        |
|   | round check the supercharger tion of the supercharger will be | control is shifted from the high to the lo indicated by | )W     |
| A) a sudden decrease                        | in manifold pressure.   |   |        |
| B) the manifold pressu                      | re remaining the same.  |   |        |
| C) a momentary increa                       | se in manifold pressure.                                      |   |        |

| 180.   | S49                     | FEX   |
|--|-------------------------|---|
| is equipped with a single-                           |                         | er control when executing a go-around if the engine percharger? |
| A) Off.  |                         |   |
| B) Low blower.                                       |                         |   |
| C) High blower.                                      |                         |   |
| 181.   | T12                     | FEX   |
| What effect does an increal alcohol injected engine? | ease in atmospheric h   | numidity have on brake horsepower output of a water             |
| A) A power loss will be ex                           | xperienced by either a  | a wet or dry takeoff.   |
| B) A wet engine takeoff w                            | vill lose power more r  | apidly than a dry engine takeoff.                               |
| C) A pressure-injected ca                            | arburetor will not be a | ffected by increased humidity.                                  |
| 182.   | T12                     | FEX   |
| What will result if too muc                          | ch antidetonation fluid | d is injected during takeoff?                                   |
| A) Power loss.                                       |                         |   |
| B) Temporary power incre                             | ease.                   |   |
| C) Engine overheat and o                             | detonation.             |   |
| 183.   | T12                     | FEX   |
| How does increased hum                               | nidity affect engine pe | rformance on takeoff?   |
| A) Engine temperatures v                             | will increase.          |   |
| B) The fuel/air ratio will de                        | ecrease below best p    | ower.   |
| C) Water vapor will displa                           | ace oxygen which inc    | reases the mixture richness.                                    |
| 184.   | T12                     | FEX   |
| How does high humidity a                             | affect engine perform   | ance on takeoff without water injection?                        |
| A) Power will increase sli                           |                         | ·   |
| B) Power will decrease so                            |                         |   |
| C) Power will not change                             | •                       |   |
| 185.   | T12                     | FEX   |
| Alcohol is added to the flu                          | uid in a water injectio | n system to   |
| A) increase the octane.                              |                         |   |
| B) provide greater cooling                           | g.                      |   |
| C) prevent freezing of the                           | e water.                |   |

| 186.                                   | T12   | FEX   |                |
|--|---|---|----------------|
| A minimum loss of p                    | ower will occur on takeoff in                                 | high humidity conditions if                                       |                |
| A) the carburetor is                   | set at full rich.   |   |                |
| B) the mixture is set                  | for emergency rich and add                                    | itional fuel is injected.   |                |
| C) the fuel/air ratio is               | s set for best power and the                                  | takeoff is wet (with antidetonation fluid).                       |                |
| 187.                                   | S48   | FEX   |                |
| The first step to corr                 | ect preignition is to   |   |                |
| A) retard the throttle                 |   |   |                |
| B) increase the mixt                   | ure.  |   |                |
| C) decrease the mix                    | ture.   |   |                |
| 188.                                   | S48   | FEX   |                |
| If an attempt is made                  | e to start an engine with a h                                 | /draulic lock,  |                |
| A) a connecting rod                    | can bend or break if the cra                                  | nkshaft continues to rotate.                                      |                |
| B) the starter gearbo                  | ox can overtorque since the                                   | liquid is incompressible and stops piston                         | movement       |
| C) the fuel or oil from                | n the lower cylinders can be                                  | injected into the exhaust system causing                          | ı afterfiring. |
| 189.                                   | S48   | FEX   |                |
| Which procedure sh                     | ould be followed prior to sta                                 | ting an engine with a hydraulic lock?                             |                |
| A) Remove the lowe                     | r sparkplugs and rotate the                                   | engine in the normal direction of rotation.                       |                |
| B) Remove the front minimum of two com |   | e top cylinders and pull the propeller throu                      | ugh a          |
|  | er sparkplugs and pull the en<br>ne intake pipe and causing a | gine through backwards to prevent fuel of lock on the next start. | or oil from    |
| 190.                                   | T09   | FEX   |                |
| What may cause en                      | gine detonation?  |   |                |
| A) High octane fuel.                   |   |   |                |
| B) Low manifold pre                    | ssure.  |   |                |
| C) High intake air te                  | mperatures.   |   |                |
| 191.                                   | S14   | FEX   |                |
| What does an increa                    | ase of approximately 125 pro                                  | ppeller RPM indicate during shutdown?                             |                |
| A) Idle mixture is con                 | rrect.  |   |                |
| B) Idle mixture is too                 | lean.   |   |                |
| C) Idle mixture is too                 | rich.   |   |                |
|  |   |   |                |

| 192.   | S54   | FEX  |
|--|---|--|
| Which is a feature of the prope<br>A) Throttle levers may be in any<br>B) Retarding a throttle to idle or<br>C) Propeller blades are held in           | y position for the autofeather at takeoff will cause the autofe | eather system to operate.                        |
| 193.<br>What initial action is taken to ur<br>A) Place the aircraft in a shallow<br>B) Turn the autofeather system<br>C) Hold the feather button in ur | w dive to start the propeller w off and place the propeller le  | indmilling.<br>ver to the full forward position. |
| 194. Retarding the throttle setting in A) a decrease in blade angle. B) an increase in blade angle. C) a decrease in propeller RPM                     | -   | FEX  |
| 195.<br>Which operational force creates<br>A) Centrifugal.<br>B) Torque bending.<br>C) Aerodynamic twisting.   | S54<br>s the greatest stress on a pro                           | FEX<br>peller?                                   |
| 196. In the propeller deicing system, A) through slip rings and carbo B) through flexible electrical co C) by use of slip rings and segr               | n brushes.<br>nnectors.   | FEX<br>d to the propeller hub assembly           |
| 197. The force which tries to feather A) torque bending. B) aerodynamic twisting. C) centrifugal twisting moment.                                      |   | FEX  |
| 198.   | S18   | FEX  |
| Which flight conditions will resu  | ılt in the largest propeller blad                               | e angle?   |
| <u>-</u>   | · · ·   | •  |

| A) Initial climb-out.                |                               |  |
|--------------------------------------|-------------------------------|--|
| B) Approach to landin                | ıg.                           |  |
| C) High-speed, high-a                | altitude cruise flight.       |  |
| 199.                                 | S18                           | FEX  |
|                                      |                               | feathering propeller set in the HIGH RPM position          |
| when the feathering a                |                               | cathorning propositor set in the File File in the position |
| A) Low pitch through                 | reverse, to feather.          |  |
| B) High pitch through                | low pitch, to feather.        |  |
| C) Low pitch through                 | high pitch, to feather.       |  |
| 200.                                 | T55                           | FEX  |
| What is the minimum                  | glycol content of Type 1 de   | eicing/anti-icing fluid?                                   |
| A) 30 percent.                       |                               |  |
| B) 50 percent.                       |                               |  |
| C) 80 percent.                       |                               |  |
| 201.                                 | T55                           | FEX  |
| Which of the following step process? | g procedures will increase t  | the holding time during the anti-ice phase of a two        |
| A) Glycol content is ra              | aised to 100 percent.         |  |
| B) The Type 2 fluid is               | heated before application.    |  |
| C) The Type 2 fluid is               | applied with centrifugal pu   | ımps.  |
| 202.                                 | M51                           | FEX  |
|                                      | ng ethylene glycol deicing fl | uid with water in nonprecipitation conditions is to        |
| A) raise the eutectic p              |                               | • •  |
| B) decrease the freez                |                               |  |
| •                                    | num freeze point (onset of o  | crystallization).  |
| 203.                                 | T55                           | FEX  |
|                                      |                               | dispensed by a ground unit?                                |
| A) Cold.                             | Ü                             | , , ,  |
| B) Heated.                           |                               |  |
| C) Ambient.                          |                               |  |
| 204.                                 | T55                           | FEX  |
|                                      |                               | leicing/anti-icing an airplane using a two- step           |
| orocess?                             | •                             | •  |

| A) Heated Type 1 fluid followed b                                | y hot Type 2 fluid.             |                                      |
|--|---------------------------------|--------------------------------------|
| C) Heated Type 1 or 2 fluid foll                                 | lowed by cold Type 1 fluid.     |                                      |
| 205.   | M51                             | FEX                                  |
| Anti-icing fluid should provide f                                | reezing point protection to     |                                      |
| A) -20 °F ambient temperature                                    |                                 |                                      |
| B) +32 °F outside temperature                                    |                                 |                                      |
| C) a freezing point no greater t                                 | han 20 °F below the ambient     | or airplane surface temperature.     |
| 206.   | T39                             | FEX                                  |
| What indicates corrosion below                                   | v an aluminum clad surface a    | ttacking the metal interior?         |
| A) White blotches.   |                                 |                                      |
| B) Greenish deposits.  |                                 |                                      |
| C) Small, dark grey lumps.                                       |                                 |                                      |
| 207.   | S38                             | FEX                                  |
| Self-locking nuts should be ins                                  | pected on preflight for         |                                      |
| A) a minimum of three full threa                                 | ads engaged by the nut.         |                                      |
| B) at least one full thread protr                                | •                               |                                      |
| C) at least two full threads prot                                | truding beyond the nut.         |                                      |
| 208.   | J11                             | FEX                                  |
| When are ATIS broadcasts up                                      | dated?                          |                                      |
| A) Only when the ceiling or vis                                  | ibility changes by a reportable | e value.                             |
| B) Every 30 minutes if weather                                   |                                 | ·                                    |
| C) Upon receipt of any official                                  | weather, regardless of conter   | nt change or reported values.        |
| 209.   | J11                             | FEX                                  |
| What effect does the barometral altitude information transmitted |                                 | pilot altimeters have on the MODE C  |
| A) Neither pilot altimeter setting                               | g has an effect.                |                                      |
| B) The pilot's indicated altitude                                |                                 | • •                                  |
| C) The copilot's indicated altitusecond transponder is selected  | • •                             | pped with two transponders and the   |
| 210.   | T55                             | FEX                                  |
| Which of the following will deci                                 | rease the holding time during   | anti-icing using a two-step process? |
|  |                                 |                                      |

| A) Apply heated Type 2                  | 2 fluid.                     |   |
|---|------------------------------|---|
| B) Decrease the water                   | content.                     |   |
| C) Increase the viscosit                | ty of Type 1 fluid.          |   |
| 211.                                    | J31                          | FEX   |
| To which position shoulare experienced? | ld the oxygen regulator b    | be set when symptoms of hypoxia or hyperventilation     |
| A) Normal.                              |                              |   |
| B) Emergency.                           |                              |   |
| C) 100 percent oxygen.                  |                              |   |
| 212.                                    | A01                          | FEX   |
| What is the CFR Part 1                  | definition of critical engi  | ine?  |
| A) The outboard engine                  | e whose failure would mo     | ost adversely affect safety.                            |
| B) The engine still runn                | ing which would most ac      | dversely affect performance if it should fail.          |
| C) The engine whose fa                  | ailure would most advers     | sely affect airplane performance or handling qualities. |
| 213.                                    | T30                          | FEX   |
| The angle between the the angle of      | chord line of the wing a     | nd the longitudinal axis of the airplane is known as    |
| A) attack.                              |                              |   |
| B) dihedral.                            |                              |   |
| C) incidence.                           |                              |   |
| 214.                                    | D15                          | FEX   |
| A flight engineer operat                | ting under CFR part 121      | must receive recurrent training on                      |
| A) normal operation of                  | the airplane flight system   | ns within the preceding 6 calendar months.              |
| B) emergency operation                  | n of all airplane flight sys | stems within the preceding 12 calendar months.          |
| C) alternate operation of               | of the airplane flight syste | ems within the preceding 24 calendar months.            |
| 215.                                    | D20                          | FEX   |
| Which flight crewmemb                   | ers may leave their stati    | on during cruising flight to perform normal duties?     |
| A) One pilot and the flig               | ght engineer together wh     | en required.  |
| B) Either pilot or the flig             | ght engineer, but only on    | e at a time.  |
| C) One pilot or the fligh               | t engineer if the flight en  | gineer station is occupied by a pilot.                  |
| 216.                                    | D20                          | FEX   |
| Which flight crewmemb                   | er nonessential convers      | ation is allowed below 10,000 feet?                     |
|   |                              |   |

| <ul><li>A) Discussing stock market rep</li></ul>  | oorts during taxi.  |  |
|---|---|--|
| B) Ordering something to drink  | from the galley while cruising  | at 8,000 feet.                                       |
| C) Confirming airplane logbook  | centries during climb when cle  | ear of the airport traffic area.                     |
| 217.  | D20   | FEX  |
|   |   | neer panel with his or her seatbelt                  |
| A) During taxi, takeoff, and land   | ding.   |  |
| B) During takeoff, climbs, desc   | ents, and landing.  |  |
| C) Whenever the airplane is in of his or her duties in connection   |   | ence is necessary for the performance plane.         |
| 218.  | D21   | FEX  |
| The function of the minimum e   | quipment list is to indicate req  | uired items which                                    |
| A) may be inoperative for a flig  |   |  |
| B) cannot be missing from the   |   | nt.  |
| ,   |   | domestic passenger scheduled flights.                |
| 219.  | D22   | FEX  |
|   |   | mestic air carrier flight conducted                  |
| A) Load manifest and flight rele  | ease.   |  |
| B) Dispatch release, load mani  | fest, and flight plan.  |  |
| C) Maintenance release, weigh   | nt and balance release, and flig  | ght plan.  |
| 220.  | D20   | FEX  |
| Who is responsible for noting r<br>A) Flight engineer.<br>B) Pilot in command.<br>C) Certificate holder or designa      |   | aircraft maintenance log?                            |
| 221.  | D17   | FEX  |
| Duty and rest period rules for one of the color of the A) not be assigned to any duty B) be relieved of all duty for at | domestic air carrier operations with the air carrier during a re least 48 hours during any 7 co | require that a flight crewmember quired rest period. |
| 222.  | D15   | FEX  |
|   |   |  |

| What are the minimum flight er<br>airplanes when common carria                              |                                  | requirements for reciprocating-powered     |
|---|----------------------------------|--|
| •   | •                                | pervision of a check airman in flight.     |
|   | ·                                | upervision of a qualified flight engineer. |
|   |                                  | upervision of a qualified pilot in         |
| 223.  | D15                              | FEX  |
| Which requirement must be me<br>air carrier flight under CFR par                            |                                  | 6 months before they can serve on an       |
| A) Line check or route check.   |                                  |  |
| B) Recurrent flight and ground  | · ·                              |  |
| C) 50 hours of flight time or a f   | light check.                     |  |
| 224.  | D11                              | FEX  |
| Which requirement applies to eand crash ax) installed in an ai                              | • • • • • •                      | inguishers, megaphones, first aid kits,    |
| A) The equipment must be clea   | arly marked to indicate its met  | hod of operation.                          |
| B) The equipment must be acc  | essible to the crew and passe    | engers regardless of location.             |
| C) The equipment located in th container.   | e passenger compartment mu       | ust be stored in a compartment or          |
| 225.  | D11                              | FEX  |
| Interior emergency exit lights s<br>A) be operable from the flight o                        | •                                | n. CFR's require that these lights must    |
| B) be armed or turned on durir  | •                                | ht operations.                             |
| C) be operable manually from  | the flightcrew station and the p | bassenger compartment.                     |
| 226.  | B15                              | FEX  |
| occurrence resulting in termina<br>A) 60 days.<br>B) 90 days.                               |                                  | a be kept in the event of an accident or   |
| C) 120 days.  |                                  |  |
| 227.  | D11                              | FEX  |
| Above which cabin altitude mu  A) All crewmembers 10,000 fee  B) All crewmembers 12,000 fee | et; all passengers 12,000 feet.  |  |

| C) All crewmembers 1                          | 10,000 feet; all passenge                             | rs 15,000 feet.  |
|---|---|--|
| 228.  | D13   | FEX  |
| If the flight engineer b                      |   | ho may perform flight engineer duties during an IFR  |
| A) Any crewmember of                          | designated by the pilot-in-                           | -command.  |
| B) A pilot crewmember                         | er, but only if flight engine                         | er certificated.   |
| C) Either pilot, but onl                      | y if qualified to perform fl                          | ight engineer functions.   |
| 229.  | D11   | FEX  |
|   | flight deck duty when op                              | ized air carrier transport airplanes carry for each erating at flight altitudes above 10,000 feet? |
| B) A minimum of 2 ho                          | • • •   |  |
| C) A minimum of 30 n                          |   |  |
| 230.  | D30   | FEX  |
| Which CFR governs a                           | airplane operations when                              | common carriage is not involved?   |
| A) CFR part 135 for c                         | argo flights.   |  |
| B) CFR part 121 for to                        | est flights.  |  |
| C) CFR part 125 for p                         | assenger flights.                                     |  |
| 231.  | D36   | FEX  |
| Which maintenance to                          | ask may a flight engineer                             | perform while operating under CFR part 125?  |
| A) Landing light replace                      | cement if there is no certi                           | ficated mechanic available.  |
| B) Remove, inspect, a                         | and replace a chip detecte                            | or if the malfunction occurs in a remote area.   |
|   |   | oropriate mechanic certificate is held, an approved writing by the certificate holder.             |
| 232.  | D36   | FEX  |
| Who may replace a se<br>Authority from CFR pa |   | erated under CFR part 91 by a Letter of Deviation  |
| A) Any person design holder to perform mai    | •   | ntenance organization arranged by the certificate  |
|   | as long as it does not invoved Certificate Holder's   | volve complex assembly operations and that task is Manual.   |
| •   | and regardless of whethe<br>ents of CFR part 125, Sub | r the Letter of Deviation Authority provides relief fron opart G.                                  |

| 233.  | A30                       | FEX  |         |
|---|---------------------------|--|---------|
| Which is the latest date examination was a first-   | •                         | ay perform duties if his or her last medical gust 25, 1999?  |         |
| A) September 30, 2000                               |                           |  |         |
| B) August 25, 2000                                  |                           |  |         |
| C) August 31, 2000                                  |                           |  |         |
| 234.  | A30                       | FEX  |         |
| Refusal by a flight enging inspector is grounds for | _                         | llcohol test results when requested by an FAA  |         |
| A) revocation of their flig                         | ht engineer and pilot c   | ertificates.   |         |
| B) suspension of their m                            | nedical certificate for a | period of up to 1 year.  |         |
| C) denial of an application                         | on for any certificate or | rating for a period of up to 5 years.  |         |
| 235.  | A30                       | FEX  |         |
| A person with a physica  A) may not perform fligh   | •                         | standards of CFR part 67 for their medical certif  | ficate  |
| B) must surrender the m                             | -                         | FAA inspector  |         |
| •   |                           | aminer before returning to flight crewmember d   | luties. |
|   |                           | animo do lo como como de la megana |         |
| 236.  | D30                       | FEX  |         |
| An airplane may be ope                              | rated under CFR part 9    | 91 while carrying passengers or cargo when   |         |
| A) the operator has a Le                            | etter of Deviation Autho  | rity issued under CFR part 125.  |         |
| B) the flight is for training                       | g, ferrying, positioning, | or maintenance purposes.   |         |
| C) there are less than 20                           | 0 seats installed in the  | airplane or the cargo weighs less than 6,000 po  | ounds   |
| 237.  | B07                       | FEX  |         |
| No person may act as a consumed alcoholic bev       |                           | der the influence of alcohol or when he or she leding  | has     |
| A) 8 hours.   |                           |  |         |
| B) 12 hours.  |                           |  |         |
| C) 24 hours.  |                           |  |         |
| 238.  | J26                       | FEX  |         |
| The accuracy of an altim                            | neter is questionable w   | hen the difference between the   |         |
| A) pilot and copilot altim                          | eters is more than 75 f   | eet in flight.   |         |
| B) known field elevation                            | and the indicated altitu  | ide is more than 75 feet.  |         |
| C) indicated altitude and                           | I the altitude reported b | y ATC is more than 100 feet.   |         |

| 239.   | H126                               | FEX                                |
|--|------------------------------------|------------------------------------|
| Zero fuel weight is defined as                       | s the                              |                                    |
| A) basic operating weight plu                        | s payload.                         |                                    |
| B) empty weight plus passen                          | gers and cargo.                    |                                    |
| C) takeoff weight minus fuel t                       | to destination and alternate.      |                                    |
|  |                                    |                                    |
| 240.   | H122                               | FEX                                |
| (Refer to figure 40.) What is t                      | he loaded CG in percent of MAC     | under operating conditions No. 1?  |
| A) 28.9 percent.                                     |                                    |                                    |
| 3) 30.5 percent.                                     |                                    |                                    |
| C) 32.9 percent.                                     |                                    |                                    |
| 241.   | H126                               | FEX                                |
|  |                                    |                                    |
| (Neier to rigure 32.) What is t<br>4) 20,500 pounds. | he maximum payload under oper      | ating conditions No. 1:            |
| 3) 21,500 pounds.                                    |                                    |                                    |
| C) 25,500 pounds.                                    |                                    |                                    |
| 5) 25,500 pourius.                                   |                                    |                                    |
| 242.   | H124                               | FEX                                |
| (Refer to figure 53.) What min                       | nimum weight of cargo must be sl   | hifted from the forward to the aft |
| cargo location to bring the Co                       | G within limits under operating co | nditions No. 1?                    |
| 4) 195 pounds.                                       |                                    |                                    |
| B) 410 pounds.                                       |                                    |                                    |
| C) 455 pounds.                                       |                                    |                                    |
| 243.   | H124                               | FEX                                |
|  | lane moves forward during retrac   |                                    |
| A) total moments will decreas                        | •                                  | uon, me                            |
| 3) total moments will increas                        |                                    |                                    |
| C) total moments will remain                         |                                    |                                    |
|  | ine same.                          |                                    |
| 244.   | H124                               | FEX                                |
| f the landing gear of an airpla                      | ane moves rearward upon gear re    | etraction, the CG will             |
| A) move aft.   |                                    |                                    |
| B) move forward.                                     |                                    |                                    |
| C) remain the same.                                  |                                    |                                    |
|  |                                    |                                    |
| 245.   | W16                                | FEX                                |
|  |                                    |                                    |

| The purpose of airplane                                  | wing dihedral angle is to           | 0  |             |
|--|-------------------------------------|--|-------------|
| A) increase lateral stabili                              | ity.                                |  |             |
| B) increase longitudinal                                 | stability.                          |  |             |
| C) increase lift coefficien                              | it of the wing.                     |  |             |
| 0.40   |                                     | 000  | FFV         |
| 246.<br><del>-</del>                                     | tat di anna la Canada (Canada di an | S69  | FEX         |
|  | id the cabin altitude be s          | set for the following landing conditions?  |             |
| Altimeter  |                                     | 30.12  |             |
| Field elev   |                                     | 6172 ft  |             |
| Airplane cabin depressu                                  |                                     | 500 ft AGL   |             |
| Cabin pressure controlle                                 | r calibrated to                     | 29.92  |             |
| A) 6,472 feet.   |                                     |  |             |
| B) 6,672 feet.   |                                     |  |             |
| C) 6,792 feet.   |                                     |  |             |
| 247.   | S69                                 | FEX  |             |
|  | minate. Which is the ma             | oin pressure differential of 10 inches Hg with aximum altitude that the airplane can be flow |             |
| A) 9,000 feet.   |                                     |  |             |
| B) 10,000 feet.  |                                     |  |             |
| C) 11,000 feet.  |                                     |  |             |
| 248.   | S69                                 | FEX  |             |
| (Refer to figure 6.) Whicl<br>pressure differential of 1 |                                     | tude that can be maintained at FL 320 with   | a           |
| A) -1,000 feet.  |                                     |  |             |
| B) +1,000 feet.  |                                     |  |             |
| C) +8,000 feet.  |                                     |  |             |
| 249.   | T31                                 | FEX  |             |
| (Refer to figure 27.) Wha                                | at is the total fuel burn fo        | or a 1,500 NM cruising flight under operating  | J           |
| A) 19,060 pounds.  |                                     |  |             |
| B) 19,200 pounds.  |                                     |  |             |
| C) 22,500 pounds.  |                                     |  |             |
| 250.   | S24                                 | FEX  |             |
| In an electrically heated                                | windshield system, wha              | t maintains normal windshield temperature?   | <b>&gt;</b> |

| A) Thermistors.                                     |                       |  |
|---|-----------------------|--|
| B) Electronic amplifiers.                           |                       |  |
| C) Thermal overheat switch                          | ches.                 |  |
| 251.  | T66                   | FEX  |
| Why is it necessary to per                          | iodically completely  | discharge and recharge a nickel-cadmium battery? |
| A) To restore electrolyte le                        | evels.                |  |
| B) To eliminate cell imbala                         | ance and loss of cap  | eacity.  |
| C) To dissolve nickel oxide                         | e formations on pos   | itive cells to restore capacity.                 |
| 252.  | T66                   | FEX  |
| What causes cell imbaland                           | ce in a nickel-cadmi  | um battery?                                      |
| A) Low temperatures.                                |                       |  |
| B) Deep rapid discharges.                           |                       |  |
| C) Constant-potential (vol                          | age) charging.        |  |
| 253.  | T66                   | FEX  |
| What condition characterize                         | zes a thermal runaw   | ay?  |
| A) Increased resistance of                          | the battery to input  | current.   |
| B) High temperature and u                           |                       |  |
| C) Continuous rising curre                          | ent and increasing be | attery temperature.                              |
| 254.  | S66                   | FEX  |
| What speed does a freque A) Engine N <sub>2</sub> . | ency meter give a di  | rect indication of?                              |
| B) Generator RPM.                                   |                       |  |
| C) CSD input speed.                                 |                       |  |
| o, conpar epeca.                                    |                       |  |
| 255.  | T72                   | FEX  |
| Which are control function                          | s of an ac generato   | r control unit?                                  |
| A) Manual paralleling.                              |                       |  |
| B) Nonessential-power rel                           | •                     |  |
| C) Generator field control                          | and indication.       |  |
| 256.  | T66                   | FEX  |
|   |                       | the separator in a nickel-cadmium battery?       |
| A) Separates positive and                           | •                     |  |
| B) Seals the cell to prever                         | it leakage of electro | lyte.  |

| C) Inhibits oxygen formed that may lead to a thermal  |   | ng, from recombining with cadmium and creating heat  |
|---|---|--|
| 257.  | T66                                     | FEX  |
| What is a function of the s  A) Inhibits thermal runawa  B) Separates negative and  C) Prevents electrolyte lea | ay.<br>d positive cells.                | ne in a nickel-cadmium battery?  |
| 258.  | S31                                     | FEX  |
| The cabin pressure control  A) compressor speed.  B) outflow valve opening.  C) pneumatic system pres           | ·                                       | ct effect upon the   |
| 259.  | T49                                     | FEX  |
| What is an advantage of a  A) Fire hazards are reduce  B) Chemical systems may  C) Reliability is improved I    | ed by eliminating o                     | xygen lines.<br>time after they are activated.   |
| 260.  | S27                                     | FEX  |
| How does the thermocoup<br>A) Heat increases electric<br>B) Heat generates a small<br>C) Heat causes expansion  | al resistance.<br>I electrical current. | on system cause the warning system to operate?   |
| •   | nce as airspeed is                      | FEX increased to allow hands-off flight. speeds in flight allowing neutral control forces. |
| C) modify the downward to   | ail load for various                    | airspeeds in flight eliminating flight control pressures.                                  |
| 262.<br>Mixing aviation gasoline w<br>A) turbine blades.<br>B) compressor blades.<br>C) inlet guide vanes.      | S04<br>vith jet fuel will effe          | FEX ct a turbine powerplant by forming deposits on the                                     |

| 263.  | S04                         | FEX   |
|---|-----------------------------|---|
| When comparing jet fu                           | uel to aviation gasoline, w | hich statement is correct?  |
| A) Both jet fuel and ga                         | soline are equally suscep   | tible to contamination.   |
| B) Jet fuel is of a high                        | er viscosity, and holds cor | ntaminants better.  |
| C) Jet fuel is of higher                        | viscosity, and will not hol | d contaminants as well as gasoline.   |
| 264.  | S42                         | FEX   |
| What is the difference                          | between Jet A and Jet A-    | 1 fuel?   |
| A) Jet A is colorless a                         | nd Jet A-1 is straw colored | d.  |
| B) Jet A has a freeze ¡                         | point of -40 °C and Jet A-  | 1 has a freeze point of -47 °C.   |
| C) Jet A-1 is a blend of                        | of gasoline and kerosene r  | made for operating at extremely low temperatures.                                     |
| 265.  | S42                         | FEX   |
| What is the difference                          | between Jet A and Jet B     | fuel?   |
| A) Jet A is colorless a                         | nd Jet B is straw colored.  |   |
| B) Jet A has a freeze ¡                         | point of -40 °C and Jet B I | nas a freeze point of -47 °C.   |
| C) Jet B is a blend of (                        | gasoline and kerosene ma    | ade for operating at extremely low temperatures.                                      |
| 266.  | T18                         | FEX   |
| With regard to the fuel consideration in refuel |                             | above the fuel in the tanks, which should be a  |
| A) Gasoline-type fuels                          | (Jet B) produce a rich mi   | xture which is easily ignited at normal temperatures.                                 |
| B) Mixing Jet A and JF range.                   | P/4 produces a fuel/vapor   | mixture that is ignitable through a wide temperature                                  |
| C) Vapor pressure of contemperatures.           | Jet A type fuel is too high | to develop an ignitable mixture at normal   |
| 267.  | S68                         | FEX   |
|   |                             | ard because of its susceptibility to freezing as it method of preventing this hazard? |
| A) Use of micromesh f                           | fuel strainers.             |   |
| B) Use of anti-icing fue                        | el additives.               |   |
| C) Adding deicing fluid                         | to the fuel.                |   |
| 268.  | S52                         | FEX   |
| If the oil cooler relief v                      | alve should stick in the op | en position, what would be the probable result?                                       |
| A) Decreased oil temp                           | erature.                    |   |
| B) Increased oil tempe                          | erature.                    |   |

| C) Pressurization of the cas   | e and oil leakage.   |                                    |
|--|--|------------------------------------|
|  |  |                                    |
| <ul><li>270.</li><li>Why should synthetic hydra</li><li>A) High evaporation rate.</li><li>B) Vapor is extremely toxic.</li><li>C) Atmospheric moisture co</li></ul>    | S63<br>ulic fluid be stored in an airtigl<br>ntamination.  | FEX<br>ht container?               |
| <ul><li>271.</li><li>How may pressure from an</li><li>A) Constant speed drive.</li><li>B) Variable-displacement put</li><li>C) In-line variable restrictor</li></ul>   | •  | FEX be regulated?                  |
| A) The locked-wheel arming B) The antiskid switch is ma  | S64 stem enables full braking action of the control of the control of the afterlands automatically disarm the analyst automatically disarm the automatically d | peeds.                             |
| <ul><li>273.</li><li>The purpose of the antiskid</li><li>A) sense wheel speed chan</li><li>B) prevent landing with the B</li><li>C) meter the brake pressure</li></ul> | ge.  | FEX el rotation.                   |
| A) Power lever retarded below.  B) One power lever retarded  | S26 Itions will cause the landing good cruise and landing gear not to idle and the landing gear on, flaps up, and power levers   | ot in transit.<br>not locked down. |

| 275.   | S46                        | FEX   |       |
|--|----------------------------|---|-------|
| Which location has the h                       | nighest gas volume in a    | turbine engine?                                 |       |
| A) Turbine outlet.                             |                            |   |       |
| B) Compressor outlet.                          |                            |   |       |
| C) Combustion chamber                          | outlet.                    |   |       |
| 276.   | S46                        | FEX   |       |
| Which location has the h                       | nighest gas pressure in    | a turbine engine?                               |       |
| A) Turbine outlet.                             |                            |   |       |
| B) Compressor outlet.                          |                            |   |       |
| C) Combustion chamber                          | · outlet.                  |   |       |
| 277.   | T24                        | FEX   |       |
| It is important to note the                    | e maximum TIT when s       | tarting a turbine engine to                     |       |
| A) determine whether the                       | e engine must be inspe     | ected or removed and overhauled.                |       |
| B) prevent operation in edistortion.           | excess of limits which m   | nay cause compressor burn-through or metal      |       |
| C) prevent dangerous te ground service equipme | •                          | elocities which may cause injury to personnel a | and   |
| 278.   | S81                        | FEX   |       |
| Which engine instrumen damage?                 | t is most likely to show   | an elevated reading if the turbine wheel has    |       |
| A) TIT.  |                            |   |       |
| B) Engine RPM.                                 |                            |   |       |
| C) Torquemeter.                                |                            |   |       |
| 279.   | S81                        | FEX   |       |
| Which engine instrumen                         | t will indicate a higher-t | han-normal reading if the compressor has dar    | mage? |
| A) Engine RPM.                                 |                            |   |       |
| B) Torquemeter.                                |                            |   |       |
| C) Turbine inlet tempera                       | ture.                      |   |       |
| 280.   | S46                        | FEX   |       |
| How does compressor R                          | RPM affect the power or    | utput of a turboprop engine?                    |       |
| A) Power increases linear                      | arly with an increase in   | compressor speed.                               |       |
| B) Efficiency increases v                      | vhen compressor blade      | tips reach Mach 1 or slightly higher.           |       |
| C) Compressor aerodyna                         | amics cause a nonlinea     | ar increase in power relative to compressor sp  | eed.  |

| 281.   | T02                         | FEX  |           |
|--|-----------------------------|--|-----------|
| Equivalent shaft horse                           | epower (ESHP) is defined    | d as   |           |
| •  | ivered to the propeller.    |  |           |
|  | ·                           | jet thrust produced by the engine.               |           |
| •  | ·                           | to the propeller shaft that is equivalent to 33, | .000 foot |
| pounds of work per m                             | -                           | , ,  |           |
|  |                             |  |           |
| 282.   | S48                         | FEX  |           |
| While starting a turbin<br>Which procedure is co | _                           | er, a hung start occurs before the starter dise  | engages   |
| A) Shut down the eng                             | jine.                       |  |           |
| B) Increase the air ve                           | locity to the starter.      |  |           |
| C) Slowly increase the                           | e power lever until the en  | gine accelerates to idle.                        |           |
|  |                             |  |           |
| 283.   | T24                         | FEX  |           |
|  | ical parameter for a turbir | ne engine during starting?                       |           |
| A) TIT.  |                             |  |           |
| B) Oil pressure.                                 |                             |  |           |
| C) Starter engagemer                             | nt time.                    |  |           |
| 284.   | S48                         | FEX  |           |
| Heat and centrifugal f                           | orce on turbine blades ca   | uses   |           |
| A) galling.                                      |                             |  |           |
| B) creeping.                                     |                             |  |           |
| C) stretching.                                   |                             |  |           |
|  |                             |  |           |
| 285.   | S48                         | FEX  |           |
| Exceeding the engine                             | temperature limitations r   | nay result in                                    |           |
| A) discoloration of the                          | compressor blades.          |  |           |
| B) rippling of the trailing                      | ng edge of the compresso    | or blades.                                       |           |
| C) hairline cracks at ri                         | ight angles to the turbine  | blade leading and trailing edges.                |           |
|  |                             |  |           |
| 286.   | S48                         | FEX  |           |
| Which action should be<br>exceed the temperatu   | -                           | y during the start sequence if the ITT attempt   | is to     |
| A) fuel.   |                             |  |           |
| B) starter.                                      |                             |  |           |
| C) ignition.                                     |                             |  |           |

| 287.   | T25                           | FEX                                      |            |
|--|-------------------------------|--|------------|
| The propeller synch                            | ronization has a limited RPN  | I range to prevent the possibility of    |            |
| A) a runaway maste                             | er engine from overspeeding   | all the engines.                         |            |
| B) overtorqueing the                           | e other engines in case the r | naster engine fails and is feathered.    |            |
| C) the other engine                            | s losing more than a limited  | amount of RPM in case the master engine  | e fails.   |
|  |                               |  |            |
| 288.   | T26                           | FEX                                      |            |
| Beta range refers to                           |                               |  |            |
|  | ange controlled by the prop l |  |            |
|  | ngle which can produce zer    |  |            |
| C) second letter of t<br>range of the throttle |                               | epresent a constant propeller RPM within | the flight |
| 289.   | M08                           | FEX                                      |            |
| What safeguard sho                             | ould be taken when using mo   | obile ground deice/anti-ice equipment?   |            |
| A) Open the air-con                            | ditioning pack valves.        |  |            |
| B) Operate the airpl                           | ane engines above idle to p   | revent flameout.                         |            |
| C) Apply fluid to the                          | lower door sills and the doo  | r bottoms prior to closing for flight.   |            |
|  |                               |  |            |
| 290.   | J26                           | FEX                                      |            |
|  | timeters be set to 29.92 duri | ng climb?                                |            |
| A) FL 180.                                     |                               |  |            |
| B) 18,000 feet AGL                             |                               |  |            |
| C) 18,000 feet MSL                             |                               |  |            |
| 291.   | 120                           | FEX                                      |            |
|  | tropopause can be identified  |  |            |
| A) jet stream is enc                           |                               |  |            |
|  | se rate makes an abrupt cha   | nae.                                     |            |
|  | to increase as altitude incre |  |            |
| ,  |                               |  |            |
| 292.   | 120                           | FEX                                      |            |
| A characteristic of the                        | he stratosphere is            |  |            |
| A) a relatively even                           | base altitude of approximate  | ely 35,000 feet.                         |            |
| B) an overall decrea                           | ase of temperature with an ir | crease in altitude.                      |            |

C) a relatively small change in temperature with an increase in altitude.

| <ul><li>293.</li><li>What is the nominal voltage rati</li><li>A) 6 volts.</li><li>B) 12 volts.</li></ul>  | T66 ng of a fully charged lead-ac             | FEX id battery containing six cells? |
|---|---|--------------------------------------|
| C) 24 volts.  |   |                                      |
| <ul><li>294.</li><li>What is the voltage of two lead-</li><li>A) 12 volts.</li><li>B) 18 volts.</li><li>C) 24 volts.</li></ul>  | T64<br>acid 12 volt batteries connec          | FEX<br>ted in series?                |
| <ul><li>295.</li><li>What type of gas is released by</li><li>A) Toxic sulfuric acid.</li><li>B) Flammable carbon dioxide.</li><li>C) Oxygen and explosive hydro</li></ul> |   | FEX<br>narging?                      |
| <ul><li>296.</li><li>Which cabin air-conditioning sys</li><li>A) Air-cycle.</li><li>B) Vapor cycle.</li><li>C) Evaporative blower.</li></ul>                              | T49<br>stem utilizes a refrigerant to o       | FEX carry away cabin heat?           |
| 297. Which statement is correct with A) Refrigerant R-12 is nontoxic B) Refrigerant R-12 changes to C) Refrigerant R-12 changes to  | to the skin. nitric acid if it comes in conta | act with water.                      |
| 298. What is the purpose of ventilating A) Keeps the overhead thermal B) Transports heat to locations C) Provides combustion air for Q                                    | switch cool.<br>where it is needed.           | FEX<br>?                             |
| 299. What will result if an insufficient  | T12 amount of ADI fluid is injected           | FEX                                  |
| vviiai vviii 165uii ii ali 1115uiiiClefii   | amount of ADI hala is injecte                 | a during takeon!                     |

| A) Temporary power i                              | ncrease.   |  |     |
|---|--|--|-----|
| B) Engine overheat ar                             | nd detonation.   |  |     |
| C) Power will remain t                            | he same if the fuel/air rati                           | o is increased.                                    |     |
| 300.  | T10  | FEX  |     |
| What is the oil viscosit                          | ty index? The oil viscosity                            | index  |     |
| A) indicates how fluid                            | an oil is at low temperatur                            | re under laboratory conditions.                    |     |
|   | •  | hange in viscosity of an oil with changes of       |     |
| C) is the weight of any<br>Petroleum Institute (A | -  | eight of an equal volume of oil from the America   | ın  |
| 301.  | S52  | FEX  |     |
|   | with Hamilton-Standard F<br>Imp from a standpipe is to | Hydromatic propellers the purpose of feeding th    | е   |
| A) minimize the amou                              | nt of oil that has to be dilu                          | ited with fuel in cold weather.                    |     |
| B) provide oil for feath                          | ering the propeller in the                             | event an oil line should break.                    |     |
| C) separate the circula<br>warmup of the engine   | •  | ling oil when the engine is started to permit a fa | ıst |
| 302.  | T10  | FEX  |     |
| The ashless in ashles                             | s dispersant refers to oil                             |  |     |
| A) crudes that are ash                            | rfree.   |  |     |
| B) additives that leave                           | e no metallic ash.                                     |  |     |
| C) that has had all ash                           | n removed in the refining p                            | process.   |     |
| 303.  | T12  | FEX  |     |
| The main fuel sump st                             | trainers are located                                   |  |     |
| A) at the carburetor in                           | let chamber.   |  |     |
| B) at the lowest point                            | in the fuel system.                                    |  |     |
| C) on the aft side of th                          | ne firewall in each nacelle.                           |  |     |
| 304.  | T12  | FEX  |     |
| What will result if too r                         | much ADI fluid is injected                             | during takeoff?                                    |     |
| A) Power loss.                                    |  |  |     |
| B) Temporary power i                              | ncrease.   |  |     |
| C) Engine overheat ar                             | nd detonation.   |  |     |
| 305.  | T10  | FEX  |     |

| The purpose of the oil                         | cooler bypass valve is to                              |  |             |
|--|--|--|-------------|
| A) bypass the oil coole                        | r when there is a blockag                              | je.  |             |
| B) control and limit the                       | lubricating oil pressure.                              |  |             |
| C) control the flow of o                       | il to the oil cooler accordi                           | ng to the temperature and viscosity of the         | ne oil.     |
| 306.   | S12  | FEX  |             |
| The total piston displace                      | cement of a reciprocating                              | engine is the                                      |             |
| A) relationship of horse                       | epower output per cubic in                             | nch of cylinder volume.                            |             |
| B) volume displaced by                         | y one piston during one re                             | evolution of the crankshaft.                       |             |
| C) volume displaced by                         | y the sum total of all pisto                           | ons during one revolution of the cranksha          | aft.        |
| 307.   | T30  | FEX  |             |
| The BMEP indicator m                           | easures the  |  |             |
| A) ratio of the shaft out                      | tput to the power develop                              | ed in the cylinders.                               |             |
| B) output shaft torque                         | and converts it to BMEP.                               |  |             |
| C) actual power output                         | to the propeller by conve                              | erting heat energy to mechanical energy            |             |
| 308.   | S51  | FEX  |             |
| Does placing the magr handle?                  | neto switches in the OFF                               | position guarantee that the propellers ar          | e safe to   |
| A) No, the only safe wa                        | ay to ensure the engines                               | will not fire accidently is to disconnect th       | ie battery. |
| B) No, the magneto sw to the ignition system.  | ritches may fail in the clos                           | sed position and current will continue to          | be supplied |
| •  | nition the magnetos oper<br>re can allow a cylinder to | rate on the principle of short-circuiting th fire. | e current   |
| 309.   | S48  | FEX  |             |
| The purpose of shutting                        | g an engine down with th                               | e mixture control at the end of the flight         | is to       |
| A) prevent an accident                         | al start.  |  |             |
| B) preclude liquid lock                        | during subsequent starts                               |  |             |
| C) assure that there is                        | no fuel in the intake syste                            | em that could result in a fire.                    |             |
| 310.   | T11  | FEX  |             |
| Power is increased by                          | the turbines in a turbocor                             | mpound engine by                                   |             |
| A) exhaust gas velocity                        | y powering blow-down tur                               | rbines geared to the crankshaft.                   |             |
| B) engine driven turbin the manifold pressure. |  | air mixture after it leaves the carburetor         | to increase |
| C) exhaust gas pressu with metered fuel from   | <b>.</b> .   | y turbines which compress the air before           | it is mixed |

| 311.   | T30                            | FEX                                      |
|--|--------------------------------|--|
|  |                                | re that exists in the cylinder of an     |
| engine during the power stroke<br>B) The maximum power outpu<br>specified RPM and manifold p | t which can be obtained from   | an engine when it is operated at a       |
| ·  | mixture in the intake manifold | d between the carburetor or internal     |
| 312.   | S12                            | FEX                                      |
| The indicated horsepower of a  | reciprocating engine is defin  | ed by the                                |
| A) computed horsepower base  | ed on engine RPM and manifo    | old pressure adjusted to sea level.      |
| B) power developed in the con  | nbustion chambers less comp    | outed friction losses within the engine. |
| C) power developed in the conengine.   | nbustion chambers without re   | ference to friction losses within the    |
| 313.   | S48                            | FEX                                      |
| Backfiring is indicated by   |                                |  |
| A) intermittent firing and low cy  | /linder temperatures.          |  |
| B) engine roughness and a su   | dden increase in cylinder hea  | d temperatures.                          |
| C) explosions from the exhaus  | t system with torching or afte | rburning.                                |
| 314.   | T12                            | FEX                                      |
| What is the effect on the fuel/a altitude without automatic mixt                             | · · ·                          | cold to a warm area at a constant        |
| A) The engine is not capable of  | of producing as much power of  | lue to the increase in air density.      |
| B) The engine is capable of product to heat expansion.                                       | oducing more power due to a    | greater volume of air which is available |
| C) The engine is not capable of causes a richer mixture.                                     | of producing as much power of  | due to a decrease in air density which   |
| 315.   | S48                            | FEX                                      |
| Afterfiring is indicated by  |                                |  |
| A) intermittent firing and low cy  | linder temperatures.           |  |
| B) engine roughness and a su   | dden increase in cylinder hea  | d temperatures.                          |
| C) explosions from the exhaus  | t system with torching or afte | rburning.                                |
| 316.   | S54                            | FEX                                      |
| (Refer to figure 11.) The prope  | ller condition depicted is     |  |

| A) onspeed.   |                                  |   |
|---|----------------------------------|---|
| B) overspeed.   |                                  |   |
| C) underspeed.  |                                  |   |
| 317.  | S54                              | FEX                                     |
| The principle which operates a                                |                                  |   |
| A) to decrease or increase the                                | •                                |   |
| B) to decrease the blade angle                                | e and counterweights to increa   | ase the blade angle.                    |
| C) and centrifugal twisting mor increase the blade angle.     | ment to decrease the blade ar    | ngle and, counterweights and springs to |
| 318.  | S18                              | FEX                                     |
| Increasing the throttle setting v                             | while taxiing with a reciprocati | ng engine will result in                |
| A) a decrease in blade angle.                                 |                                  |   |
| B) an increase in blade angle.                                |                                  |   |
| C) an increase in propeller RP                                | M.                               |   |
| 319.  | S54                              | FEX                                     |
| The purpose of the propeller s                                | ynchrophase system is to set     |   |
| A) all propellers at exactly the                              | same RPM.                        |   |
| B) the propeller blade angles f                               | or all propellers in the same re | elative position.                       |
| C) the angular difference in the blades of the master.        | e plane of rotation between the  | e blades of the slave engines and the   |
| 320.  | S54                              | FEX                                     |
| Feathering of a Hamilton-Standa) pushing in the feather butto |                                  | n be accomplished by                    |
| B) pulling the fire emergency of                              |                                  |   |
| C) moving the propeller contro                                |                                  |   |
| 321.  | W12                              | FEX                                     |
| Which factor has the effect of i                              |                                  | TEX                                     |
| A) Dry cold air.  | noredaing vi apecu:              |   |
| B) High takeoff gross weight.                                 |                                  |   |
| C) Slush or standing water on                                 | the runway.                      |   |
| 322.  | T33                              | FEX                                     |
|   |                                  | e upper surface of the wing will be     |
| zamig mgm with zoro angle of                                  | attack, the process along the    | appor danaco or the wing will be        |

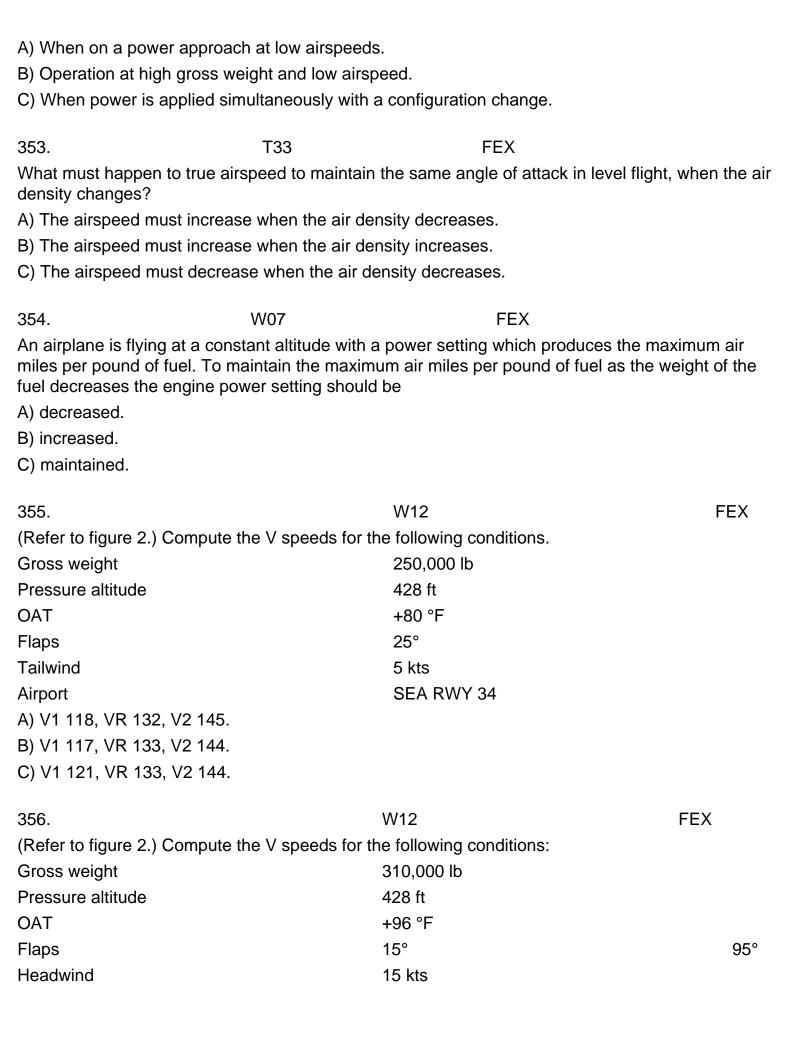
| A) equal to atmospheric pres    | sure.                        |  |
|---------------------------------|------------------------------|--|
| B) less than atmospheric pres   | ssure.                       |  |
| C) greater than the pressure    | below the wing.              |  |
|                                 |                              |  |
| 323.                            | S33                          | FEX  |
| Why will an engine produce r    | nore power on a dry day tha  | an on a humid day?                         |
| A) Increased air density.       |                              |  |
| B) Humidity increases the de    | •                            |  |
| C) Because a molecule of wa     | ater weighs more than a mol  | lecule of nitrogen or oxygen.              |
| 324.                            | T31                          | FEX  |
| (Refer to figure 26.) How mud   | ch fuel remains after dumpir | ng under operating conditions No. 1?       |
| A) 4,540 pounds.                |                              |  |
| B) 4,980 pounds.                |                              |  |
| C) 5,100 pounds.                |                              |  |
|                                 |                              |  |
| 325.                            | T31                          | FEX  |
| (Refer to figures 20 and 21.)   | What is the takeoff power a  | vailable under operating conditions No. 1? |
| A) 3,710 shaft horsepower.      |                              |  |
| B) 3,770 shaft horsepower.      |                              |  |
| C) 4,000 shaft horsepower.      |                              |  |
| 326.                            | T31                          | FEX  |
|                                 | _                            | e required for takeoff under operating     |
| conditions No. 1?               | what is the minimum torque   | required for takeon under operating        |
| A) 12,400 inch-pounds.          |                              |  |
| B) 16,600 inch-pounds.          |                              |  |
| C) 18,000 inch-pounds.          |                              |  |
|                                 |                              |  |
| 327.                            | T31                          | FEX  |
| (Refer to figure 27.) What is t | he total fuel burn under ope | erating conditions No. 1?                  |
| A) 12,800 pounds.               |                              |  |
| B) 14,440 pounds.               |                              |  |
| C) 22,160 pounds.               |                              |  |
| 328.                            | T66                          | FEX  |
| What type of gas is released    |                              |  |
| A) Toxic nickel hydroxide.      | a, a monor cadimain battor,  | ,  |
| i, iono monor ny aromao.        |                              |  |

| B) Oxygen and explosi    | ve hydrogen.  |  |  |
|--------------------------|---|--|--|
| C) Highly combustible    | acetylene and oxygen.                                 |  |  |
| 329.                     | T75   | FEX  |  |
| What is a purpose of el  | lectrical bonding jumpers                             | ?  |  |
| A) Prevent static electr | icity discharges between                              | parts of the structure.  |  |
| B) Provide a high-resis  | tance path for electrical e                           | equipment to reduce radio interference.  |  |
| C) Minimize electrolytic | c corrosion by connecting                             | dissimilar metals to form an integral unit.  |  |
| 330.                     | T75   | FEX  |  |
| What is the purpose of   | a fuse-type current limite                            | r?   |  |
| A) Prevent overloads in  | n low power circuits.                                 |  |  |
| B) Fast blow design pro  | events damage to sensiti                              | ve circuits or equipment.  |  |
| C) Permit short periods  | s of overload before the fo                           | use link melts and breaks the circuit.   |  |
| 331.                     | T45   | FEX  |  |
|                          |   | clude trailing edge tabs on the rudder and ailerend right wing up position, the right aileron trim t |  |
| A) up, and the rudder to | ab will move to the left.                             |  |  |
| •                        | ab will move to the right.                            |  |  |
| C) down, and the rudde   | er tab will move to the lef                           | t.   |  |
| 332.                     | S17   | FEX  |  |
| Oil extracts the most he | eat from which turbine en                             | gine components?   |  |
| A) Turbine bearings.     |   |  |  |
| B) Compressor bearing    | gs.   |  |  |
| C) Accessory drive bea   | arings.   |  |  |
| 333.                     | <b>S</b> 63   | FEX  |  |
| Which hydraulic operat   | tion normally uses a doub                             | ble-acting, balanced linear actuator?  |  |
| A) Brakes.               |   |  |  |
| B) Landing gear.         |   |  |  |
| C) Automatic pilot serv  | 0.  |  |  |
| 334.                     | S26   | FEX  |  |
|                          | gear position warning sys<br>ers are retarded and the | tem provide a warning in the cockpit? gear is in transit.  |  |
| -                        |   |  |  |

| B) When the power lev                          | ers are advanced and the     | e speed brakes are not retracted.          |             |
|--|------------------------------|--|-------------|
| C) When one power lev                          | ver is retarded below cruis  | se and the gear is down and locked.        |             |
| 335.   | <b>S</b> 79                  | FEX  |             |
| Why are pneumatic sta                          | arters used on most large    | turbine engines?                           |             |
| A) Less weight.                                |                              |  |             |
| B) Simple design requi                         | res no clutch.               |  |             |
| C) Air starters are mec                        | hanically more reliable that | an electrical starters.                    |             |
| 336.   | S46                          | FEX  |             |
| How does compressor                            | RPM affect the power ou      | tput of a turbofan engine?                 |             |
| A) Power increases line                        | early with an increase in c  | compressor speed.                          |             |
| B) Efficiency increases                        | when compressor blade        | tips reach Mach 1 or slightly higher.      |             |
| C) Compressor aerody                           | namics cause a nonlinear     | r increase in power relative to compresso  | r speed.    |
| 337.   | S42                          | FEX  |             |
| Which is the most effect                       | ctive extinguishing agent f  | for use on an electrical fire?             |             |
| A) Carbon dioxide.                             |                              |  |             |
| B) Methyl bromide.                             |                              |  |             |
| C) Carbon tetrachloride                        | e (Halon 04).                |  |             |
| 338.   | D14                          | FEX  |             |
| The air carrier must give                      | e instruction on such sub    | jects as respiration, hypoxia, and decomp  | oression to |
| each crewmember on p                           | pressurized airplanes ope    | rated above                                |             |
| A) 10,000 feet.                                |                              |  |             |
| B) 12,000 feet.                                |                              |  |             |
| C) 25,000 feet.                                |                              |  |             |
| 339.   | D23                          | FEX  |             |
| A crewmember certification aircraft engaged in | ate may be issued by the     | FAA to flight crewmembers on U.S. regis    | tered       |
| A) international air com                       | imerce.                      |  |             |
| B) flight crewmember t                         | raining only.                |  |             |
| C) supplemental air ca                         | rrier operations.            |  |             |
| 340.   | D22                          | FEX  |             |
| Among the required ite                         | ms of information on the     | dispatch release of a domestic air carrier | is the      |
| A) minimum fuel supply                         | <b>y</b> .                   |  |             |

| B) weight and balance                             | e data.                        |  |      |
|---|--------------------------------|--|------|
| C) name of the pilot in                           | n command.                     |  |      |
| 341.  | B15                            | FEX  |      |
| For what purpose manot be used?                   | y information obtained from    | n cockpit voice recorders and flight data record | ers  |
| A) Identifying malfund                            | ctions and irregularities in a | aircraft systems.                                |      |
| B) Determining cause<br>Transportation Safety     |                                | ences under investigation by the National        |      |
| C) Determining any c                              | ertificate action or civil per | alty arising out of an accident or occurrence.   |      |
| 342.  | D11                            | FEX  |      |
| When a second mega A) On the flight deck.         | aphone is required, where      | will it be located?                              |      |
| B) At the forward end                             | I of the passenger compart     | ment.  |      |
| C) At the most rearwa                             | ard location in the passeng    | er compartment.                                  |      |
| 343.  | D11                            | FEX  |      |
| Which airplanes mus                               | t be equipped with flight re   | corders?   |      |
| A) Large airplanes ce                             | ertified for operations above  | e FL 250 or is turbine-engine powered.           |      |
| B) Airplanes that carr                            | ry more than 19 passenger      | S.   |      |
| C) All airplanes opera                            | ating under CFR parts 121      | or 125.  |      |
| 344.  | D30                            | FEX  |      |
| During what situation A) Test flight.             | may an airplane requiring      | a flight engineer be operated under CFR part 9   | 1?   |
| B) Cargo flight.                                  |                                |  |      |
| C) Passenger flight w                             | vithout compensation.          |  |      |
| 345.  | M51                            | FEX  |      |
| Which is an effect of                             | ice, snow, or frost formatio   | n on an airplane?                                |      |
| A) Increased stall spe                            | ed.                            |  |      |
| B) Increased pitch-do                             | own tendencies.                |  |      |
| C) Increased angle o                              | f attack for stalls.           |  |      |
| 346.  | 122                            | FEX  |      |
| What is the relationship flying at 15,000 feet in | -                              | the altimeter setting is higher than standard wh | nile |
| A) Indicated altitude i                           | s higher than true altitude.   |  |      |

| B) Indicated altitude is   | s lower than pressure altitu | ıde.   |
|----------------------------|------------------------------|--|
| C) Indicated altitude is   | s higher than pressure altit | ude.   |
| 347.                       | 122                          | FEX  |
| If the OAT increases of    | during a flight at a constan | t power and indicated altitude, the true airspeed wil                        |
| A) decrease and true       | altitude will decrease.      |  |
| B) increase and true a     | ıltitude will decrease.      |  |
| C) increase and true a     | altitude will increase.      |  |
| 348.                       | H124                         | FEX  |
| (Refer to figure 36.) W 1? | hat is the new CG after re   | emoving the weight under operating conditions No.                            |
| A) 25.3 percent.           |                              |  |
| B) 27.8 percent.           |                              |  |
| C) 31.1 percent.           |                              |  |
| 349.                       | H124                         | FEX  |
| (Refer to figure 32.) W 1? | hat is the new CG after re   | emoving the weight under operating conditions No.                            |
| A) 21.8 percent.           |                              |  |
| B) 22.4 percent.           |                              |  |
| C) 28.8 percent.           |                              |  |
| 350.                       | H124                         | FEX  |
| cargo location to bring    |                              | argo must be shifted from the forward to the aft operating conditions No. 1? |
| A) 750 pounds.             |                              |  |
| B) 1,110 pounds.           |                              |  |
| C) 1,230 pounds.           |                              |  |
| 351.                       | H124                         | FEX  |
| •                          |                              | argo must be shifted from the forward to the aft operating conditions No. 1? |
| A) 500 pounds.             |                              |  |
| B) 2,372 pounds.           |                              |  |
| C) 4,990 pounds.           |                              |  |
| 352.                       | W13                          | FEX  |
| When will power applied    | cations cause the greates    | t change in airplane trim and stability?                                     |



| Airport A) V1 139, VR 157, V2 166. B) V1 143, VR 155, V2 166. C) V1 141, VR 156, V2 165.                                   | SEA RV                         | VY 16   |     |
|--|--------------------------------|---|-----|
| 357. (Refer to figure 6.) The cabin power what is the cabin pressure altitude A) 4,200 feet. B) 4,800 feet. C) 5,010 feet. |                                | FEX PSI and the airplane is flying at FL 34   | 0.  |
| 358.   |                                | T31   | FEX |
|  | conditions?  ng and descent to | each maximum landing weight at  3 171,000 lb 142,500 lb 3,170 lb/hr/eng 19 min 2,300 lb/min |     |
| 359.   |                                | T31   | FEX |
|  | approximate duration of t      | he passenger oxygen system for the 15,000 ft 120 1,500 PSI                                  |     |
| •  |                                |   |     |
| C) 25 minutes.   |                                |   |     |
| 360. (Refer to figure 3.) What is the a conditions shown?  | approximate duration of t      | T31<br>he passenger oxygen system for the   | FEX |

| Cabin altitude  |                                 | 19,000 ft                          |
|---|---------------------------------|------------------------------------|
| Passengers  | Ļ                               | 55                                 |
| Bottle pressure   | •                               | 1,300 PSI                          |
| A) 35 minutes.  |                                 |                                    |
| B) 42 minutes.  |                                 |                                    |
| C) 46 minutes.  |                                 |                                    |
| 361.  | M51                             | FEX                                |
| What is the lowest temperature                                  | e that water droplets may rem   | ain in a liquid state?             |
| A) 0 °C.  |                                 |                                    |
| B) 0 °F.  |                                 |                                    |
| C) -40 °C.  |                                 |                                    |
| 362.  | S31                             | FEX                                |
| When the cabin pressure regu<br>vented to the atmosphere by the |                                 | ential mode, reference pressure is |
| A) relief valve.  |                                 |                                    |
| B) isobaric metering valve.                                     |                                 |                                    |
| C) differential metering valve.                                 |                                 |                                    |
| 363.  | S31                             | FEX                                |
| Which component gives an incomeasurement is used?               | dication of the rate of change  | in cabin altitude and what unit of |
| A) Pressure controller, PSI.                                    |                                 |                                    |
| B) Cabin vertical-velocity indic                                | ator, PSI.                      |                                    |
| C) Cabin vertical-velocity indic                                | ator, feet per minute.          |                                    |
| 364.  | T49                             | FEX                                |
| Which type of oxygen system                                     | is the flight deck equipped wit | h normally?                        |
| A) Constant-flow.   |                                 |                                    |
| B) Phase dilution.  |                                 |                                    |
| C) Diluter-demand.  |                                 |                                    |
| 365.  | T49                             | FEX                                |
| What type of oxygen system is                                   | s used for passengers?          |                                    |
| A) Demand.  |                                 |                                    |
| B) Constant-flow.   |                                 |                                    |
| C) Diluter-demand.  |                                 |                                    |

| 366.  | S27                             | FEX                                       |
|---|---------------------------------|---|
| How does a photoelectric smol                                     | ke detector operate?            |   |
| A) A photoelectric smoke detec                                    | ctor only warns when smoke i    | s present.                                |
| B) A photoelectric smoke detec                                    | ctor measures the amount of     | smoke under a specific set of conditions. |
| C) A photoelectric smoke detec                                    | ctor measures the amount of     | light available under a specific set of   |
| conditions.   |                                 |   |
|   | 00-                             |   |
| 367.  | S27                             | FEX                                       |
| From a standpoint of toxicity ar<br>turbojet airplanes?           | nd corrosion hazard, which fire | e extinguishing agent is safest to use in |
| A) Carbon dioxide.  |                                 |   |
| B) Methyl bromide.  |                                 |   |
| C) Chlorobromomethane.  |                                 |   |
| of officionation and  |                                 |   |
| 368.  | S53                             | FEX                                       |
| On a built-in carbon dioxide fire                                 | e extinguishing system, how is  | s a thermal discharge detected?           |
| A) The thermal plug is missing                                    | from the side of the bottle.    | -   |
| B) The red plastic disc in the th                                 | ermal discharge line is missir  | ng.                                       |
| C) The yellow plastic disc in the                                 | -                               |   |
|   |                                 |   |
| 369.  | S82                             | FEX                                       |
| When an airplane is equipped common cause of false fire wa        | •                               | tection system, which is the most         |
| A) Moisture in the system.  |                                 |   |
| B) Dents, kinks, or crushed ser                                   | nsors.                          |   |
| C) Improper routing or clampin                                    | g of detector loops.            |   |
|   |                                 |   |
| 370.  | S53                             | FEX                                       |
| In some fire extinguishing syste<br>indicated by the absence of a | ems, evidence that the systen   | n has been intentionally discharged is    |
| A) red disc on the side of the fu                                 | uselage.                        |   |
| B) green disc on the side of the                                  | e fuselage.                     |   |
| C) yellow disc on the side of th                                  | e fuselage.                     |   |
|   |                                 |   |
| 371.  | S27                             | FEX                                       |
| Which type of fire detector circondetector loops?                 | uit can continue to function wi | th either one open or one short in the    |
| A) Continuous loop.   |                                 |   |
|   |                                 |   |

| B) I wo-wire thermal switch. C) Single-wire thermal switch.  |   |  |
|--|---|--|
| 372.   | S55   | FEX  |
| The purpose of vortex generat  A) decrease drag at high airsp  B) increase elevator effectiven  C) prevent flow separation ove | eeds.<br>ess at high speeds.                                      |  |
| 373.   | T45   | FEX  |
|  | ne event of manual reversion.<br>flecting in the proper direction | to move a primary flight control.  position due to aerodynamic forces. |
| 374.<br>The speed (RPM or percent) o<br>referred to as<br>A) N <sub>1</sub> .  | S46 If the innermost compressor of                                | FEX  If a triple-spool turbofan engine is                              |
| B) N <sub>2</sub> .  |   |  |
| C) N <sub>3</sub> .  |   |  |
| 375.<br>(Refer to figure 9.) Combustior<br>A) Location 2.<br>B) Location 4.<br>C) Location 7.                                  | S73<br>n takes place in which location                            | FEX<br>n?  |
| 376.<br>(Refer to figure 9.) The total-pr  | S73<br>ressure (P <sub>t7</sub> ) probes are mour                 | FEX nted in which location?  |
| A) Location 1. B) Location 3. C) Location 10.  |   |  |
| 377.<br>Which is a disadvantage of the<br>airplane?  | T55<br>e one-step over the two-step p                             | FEX process when deicing/anti-icing an                                 |

| A) It is more complicated.                                     |                                |   |
|--|--------------------------------|---|
| B) The holding time is increase                                | sed.                           |   |
| C) More fluid is used with the flushed off airplane surfaces.  | one-step method when large     | deposits of ice and snow must be            |
| 378.   | M08                            | FEX   |
| On most airplanes, deicing of                                  | the                            |   |
| A) fuselage should be from th cracking or crazing.             | e bottom up to prevent therm   | al shock to windows which may result in     |
| B) tail surfaces should be from snow deposits into the balance |                                | reduce the possibility of flushing ice or   |
| C) wings should begin at the increasing the snow load on o     |                                | ng in an aft and inboard direction to avoid |
| 379.   | T55                            | FEX   |
| What determines the viscosity                                  | of Type 1 deicing/anti-icing t | fluid?                                      |
| A) Temperature.  |                                |   |
| B) Thickening agents.  |                                |   |
| C) Dispensing equipment.                                       |                                |   |
| 380.   | J23                            | FEX   |
| Which frequency is preferred A) 121.5 Mhz VHF.                 | to declare an emergency to A   | ATC?  |
| B) 243.0 Mhz UHF.  |                                |   |
| C) The one in use.   |                                |   |
| 381.   | J23                            | FEX   |
| Which transponder code mea                                     | ns the airplane is being force | d to a new destination?                     |
| A) 7500.   |                                |   |
| B) 7600.   |                                |   |
| C) 7700.   |                                |   |
| 382.   | S42                            | FEX   |
| What identifies a fire extinguis                               | sher used for brake fires?     |   |
| A) A square with the letter B.                                 |                                |   |
| B) A circle with the letter C.                                 |                                |   |
| C) A star with the letter D.                                   |                                |   |
| 383.   | S42                            | FEX   |

| Which is the preferred method<br>A) Spray with Halon 1301.                    | of extinguishing a brake fire of | on the ground?   |
|---|----------------------------------|--|
| B) Apply a dry powder extingui C) Blanket the fire with Halon                 |                                  |  |
| 384.<br>Which is a definition of $V_2$ spec                                   | A02<br>ed?                       | FEX  |
| A) Takeoff safety speed. B) Minimum takeoff speed. C) Takeoff decision speed. |                                  |  |
| 385.  | D14                              | FEX  |
|   | Boeing 727-100) before they      | pers who have qualified and served on may serve in the same capacity on a  |
| 386.  | D14                              | FEX  |
| Which flight engineer applicant simulator?                                    | ts may complete the entire init  | tial flight check in an approved   |
| ,   | ,                                | ht check in an airplane similar in type. n instrument, category, and class |
| C) Applicants who have flown check is being conducted.                        | more than 200 hours as a fligl   | nt engineer in the aircraft over which the                                 |
| 387.  | D20                              | FEX  |
| The pilot in command has ememany be excluded from this are                    |                                  | eople from the flight deck. Those who                                      |
| A) any person, in the interest o  | f safety.                        |  |
| B) anyone except an FAA air c   | •                                |  |
| C) anyone except a Federal la   | w enforcement officer who pre    | esents proper credentials.   |
| 388.  | D15                              | FEX  |
| What are the minimum flight er<br>airplanes when common carria                |                                  | requirements for turbojet-powered  |
| A) Flight engineer duties perfo   | rmed for 12 hours under the s    | upervision of a check airman.  |

| B) Flight engineer duties   | performed for 8 hour      | s under the supervision of the pilot in command.            |
|---|---------------------------|---|
| C) Flight engineer duties   | performed for 10 hou      | irs under the supervision of a qualified flight engineer.   |
| 389.  | D18                       | FEX   |
| What is the flight time lin additional flight crewmer                         |                           | ions that requires two pilots and at least one              |
| A) 100 hours during any   | 30-day period.            |   |
| B) 300 hours during any   | 3 calendar months.        |   |
| C) 1,000 hours during ar  | ny 12 calendar-month      | period.   |
| 390.  | D11                       | FEX   |
| On airplanes requiring a regard to the instrument                             |                           | -and-pitch indicator, which is a requirement with ?         |
| A) The power source mu<br>automatic power transfe                             |                           | ed to prevent an inadvertent failure during an              |
| B) The power source mu<br>electrical generating sys                           | •                         | eration for 30 minutes after total failure of the           |
| C) The power source mu<br>alternating current electr                          | •                         | eration for the duration of the flight after failure of the |
| 391.  | D11                       | FEX   |
| Which factors must be re  | ecorded by the approv     | red flight recorder?  |
| A) Airspeed, time, altitud  | le, vertical acceleratio  | n, and heading.   |
| B) Time, true altitude, ca  | ılibrated airspeed, ver   | tical speed, and heading.                                   |
| C) Elapsed time, airspee  | ed, altitude, vertical ac | celeration, and magnetic course.                            |
| 392.  | D11                       | FEX   |
| A flight crewmember mu  | st be able to don and     | use a quick-donning type oxygen mask within                 |
| A) 5 seconds.   |                           |   |
| B) 10 seconds.  |                           |   |
| C) 15 seconds.  |                           |   |
| 393.  | A31                       | FEX   |
|   |                           | ance of a flight engineer certificate?                      |
| <ul><li>A) Be a high school grad</li><li>B) Hold a first- or second</li></ul> | ·                         | cato  |
| •   |                           | anguage without impediment or accent.                       |
| 394.  | 129                       | FEX   |

| Why is frost considered a fl                   | ight hazard?                          |  |       |
|--|---------------------------------------|--|-------|
| A) Frost changes the basic                     | aerodynamic shape                     | of the airfoil.                                  |       |
| B) The increased weight re                     | quires a greater take                 | off distance.                                    |       |
| C) Frost causes early airflo                   | w separation resulting                | g in a loss of lift.                             |       |
|  |                                       |  |       |
| 395.   | 120                                   | FEX  |       |
| Which is true concerning th                    | e troposphere?                        |  |       |
| <ul><li>A) It extends to a uniform h</li></ul> | eight at all latitudes.               |  |       |
| B) It is thicker over the Equ                  | ator than over the po                 | les.   |       |
| C) It is the dividing line between             | ween the stratosphere                 | e and the atmosphere.                            |       |
| 396.   | T33                                   | FEX  |       |
| What atmospheric condition                     | n will decrease air de                | nsity?   |       |
| A) Decreasing humidity.                        |                                       |  |       |
| B) Decreasing pressure.                        |                                       |  |       |
| C) Decreasing temperature                      | ).                                    |  |       |
|  |                                       |  |       |
| 397.   | J26                                   | FEX  |       |
| What will be the approxima                     | te altimeter indication               | after failing to reset the local barometric pre- | ssure |
| of 30.57 after descending f                    | rom FL 250 to a field                 | elevation of 650 feet?                           |       |
| A) Sea level.                                  |                                       |  |       |
| B) 715 feet.                                   |                                       |  |       |
| C) 1,300 feet.                                 |                                       |  |       |
| 398.   | 122                                   | FEX  |       |
| Under what condition is pre                    | essure altitude and de                | ensity altitude the same value?                  |       |
| A) At standard temperature                     |                                       | ·  |       |
| B) When the altimeter setti                    | ng is 29.92 inches Ho                 | <b>J.</b>  |       |
|  | -                                     | e same value on the altimeter.                   |       |
| ,  |                                       |  |       |
| 399.   | H111                                  | FEX  |       |
| The CG of an airplane is no                    | ormally located in the                | fuselage at a point expressed in                 |       |
| A) inches from the forward                     | CG limit.                             |  |       |
| B) percent of mean aerody                      | namic chord aft of LE                 | MAC.   |       |
| C) percentage of MAC aft of                    | of the leading edge of                | the wing.  |       |
| 400.   | H111                                  | FEX  |       |
| The term mean aerodynam                        | nic chord may be defir                | ned as the                                       |       |
| -  | · · · · · · · · · · · · · · · · · · · |  |       |

- A) ratio of the average wing chord to its aerodynamic center of pressure. B) distance from the leading edge to the trailing edge of the wing, measured at the wing root. C) chord of an imaginary airfoil which has the same aerodynamic characteristics as the actual airfoil. 401. **FEX** H126 What is the maximum payload under these conditions? Basic operating weight 100,500 lb Max. zero fuel weight 138,000 lb Max. landing weight 142,000 lb Max. takeoff weight 184,200 lb Fuel tank load 54,000 lb Est. fuel burn en route 40,000 lb A) 27,500 pounds. B) 30,500 pounds. C) 33,000 pounds. **FEX** 402. H126 What is the maximum payload under these conditions? Basic operating weight 150,000 lb Max. zero fuel weight 230,000 lb Max. landing weight 245,000 lb Max. takeoff weight 320,000 lb Fuel tank load 94,500 lb Est. fuel burn en route 71,500 lb A) 72,000 pounds. B) 80,000 pounds. C) 84,000 pounds. 403. H124 FEX What is the location of the CG if 1,460 pounds are removed from Station 1500? 171,520 lb Aircraft weight Station 820 CG location A) 814.17 inches.
- 404. H105 FEX

Based on this information, where would the CG be located?

B) 850.49 inches.C) 1,000.6 inches.

Weight No. 1 601 lb at 45 in. aft of datum
Weight No. 2 700 lb at 145 in. aft of datum
Weight No. 3 125 lb at 185 in. aft of datum

A) 100.06 inches aft of datum.

B) 106.36 inches aft of datum.

C) 116.26 inches aft of datum.

405. H124 FEX

May 1,000 pounds of baggage be shifted from Station 30.0 to Station 120.0 without exceeding the aft CG limit?

Total weight 147,500 lb

CG location Station 115.8
Aft CG limit Station 118.0

- A) Yes, the CG would be located at Station 115.19.
- B) No, the new CG would be located at Station 118.41.
- C) Yes, the new CG would be located at Station 116.41.